



U.S. Department of
Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020
2/28/2011

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See Title 14 CFR §43.9, Part 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty for each such violation. (49 U.S.C. §46301(a))

1. Aircraft	Nationality and Registration Mark N67HY	Serial No. 2065	
	Make Aviat	Model A-1B	Series
2. Owner	Name (As shown on registration certificate) GARRAI LLC		Address (As shown on registration certificate)
			Address 15 Lovell St City Somers State NY Zip 10589 Country USA

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial Number
<input type="checkbox"/>	<input checked="" type="checkbox"/>	AIRFRAME	_____	(As described in Item 1 above)	_____
<input type="checkbox"/>	<input type="checkbox"/>	POWERPLANT			
<input type="checkbox"/>	<input type="checkbox"/>	PROPELLER			
<input type="checkbox"/>	<input type="checkbox"/>	APPLIANCE	Type		
			Manufacturer		

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency		
Name Tally-Ho Aviation	<input checked="" type="checkbox"/> U.S. Certificated Mechanic	<input type="checkbox"/> Manufacturer	
Address 53 Miry Brook Road	<input type="checkbox"/> Foreign Certificated Mechanic	C. Certificate No.	
City Danbury State CT	<input type="checkbox"/> Certificated Repair Station	3121251	
Zip 06810 Country USA	<input type="checkbox"/> Certificated Maintenance Organization		

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B <input type="checkbox"/>	Signature/Date of Authorized Individual 7/3/2017
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7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

BY	<input type="checkbox"/> FAA Fit Standards Inspector	<input type="checkbox"/> Manufacturer	<input type="checkbox"/> Maintenance Organization	<input type="checkbox"/> Person Approved by Canadian Department of Transport
	<input type="checkbox"/> FAA Designee	<input type="checkbox"/> Repair Station	<input checked="" type="checkbox"/> Inspection Authorization	Other (Specify)

Certificate or Designation No. 3121251	Signature/Date of Authorized Individual 07/03/2017
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NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N67HY

Nationality and Registration Mark

July 3, 2017

Date

Installed Rosen visors per STC SA01251SE

STC SA01251SE was issued to Rosen Sunvisor Systems LLC on August 6, 2003 and ammended December 8, 2008 and February 28, 2012.

Installation was accomplished in accordance with Rosen Sunvisor Sytems installation instructions for Aviat Husky A-1, A-1A, A1-B NSA Sunvisor Systems with tube mount installation.

All proper logbook entries have been made.

This modification has no other obvious effects on any other aircraft systems.

Weight and balance change is negligable.

Equipment list has been updated to reflect this change. Instructions for continued airworthiness can be found on in the installation instructions page 3.

END

Additional Sheets Are Attached

United States of America
Department of Transportation—Federal Aviation Administration
Supplemental Type Certificate

Number SA01251SE

This certificate, issued to: Rosen Sunvisor Systems, LLC
86365 College View Road
Eugene, OR 97405

*certifies that the change in the type design for the following product with the limitations and conditions therefor as specified herein meets the airworthiness requirements of Part 21 of the Federal Aviation Administration Regulations. *Certification basis established in Type Certificate Data Sheet.*

Original Product—Type Certificate Number: * See attached FAA Approved Model
Make: List (AML) No. SA01251SE for a list of approved models
Model: and applicable airworthiness regulations.

Description of the Type Design Change:

The manufacture of a Sunvisor System in accordance with Rosen Sunvisor Systems, LLC Master Drawing List defined on attached AML.

Limitations and Conditions:

1. Compatibility of this design change with previously approved modifications must be determined by the installer.
2. The Sunvisor System must be installed and maintained in accordance with the Component Maintenance Manual With Illustrated Parts List document as noted on attached AML.
3. A copy of this certificate and FAA Approved Model List SA01251SE must be maintained as part of the permanent records of the aircraft.
4. If the holder agrees to permit another person to use this certificate, the holder shall give the other person written evidence of that permission.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: May 14, 2003

Date reissued:

Date of issuance: August 6, 2003

Date amended: December 8, 2008 February 28, 2012



By direction of the Administrator



(Signature)
Greg Johnson, Acting Manager
Denver Aircraft Certification Office

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.

FAA Approved Model List (AML) No. SA01251SE
Rosen Sunvisor Systems, LLC

For

Installation of Rosen Aircraft Sunvisor Systems

Issue Date: August 6, 2003


Item	Aircraft Make	Aircraft Model	Original Type Certificate Number	Certification Basis for Alteration	Drawing List	Installation and Continued Airworthiness Instructions	AML Amendment Date
1	Maule	Bee Dee M-4, M-4C, M-4S, M-4T, M-4-180C, M-4-180S, M-4-180T, M-4-210, M-4-210C, M-4-210S, M-4-210T, M-4-220, M-4-220C, M-4-220S, M-4-220T, M-5-180C, M-5-200, M-5-210C, M-5-210T, M-5-220C, M-5-235C, M-6-180, M-6-235, M-7-235, MX-7-235, MX-7-180, MT-235, M-8-235, MX-7-160, MX-7-180A, MX-7-180B, MX-7-420, MX-7-180B, M-7-235B, M-7-235A, M-7-235C, MX-7-180C, M-7-260, MT-7-260, M-7-260C, M-7-420AC, MX-7-160C, MX-7-180AC, M-7-420A, MT-7-420, MX-7-180	3A23	Title 14 CFR 23	R1710000-DL, Revision I, dated March 10, 8/26/08 or later FAA approved revision	Document 9051-0171-001, Revision B, dated March 10, 2008 and Document 9051-0171-002, Revision A, dated June 11, 2008; or later FAA accepted revision.	12/8/08
2	Aviat	Husky A-1, A-1A, A-1B	A22NM	Title 14 CFR 23	R1710000-DL, Revision I,	Document 9051-0171-001, Revision B,	12/8/08

Item	Aircraft Make	Aircraft Model	Original Type Certificate Number	Certification Basis for Alteration	Drawing List	Installation and Continued Airworthiness Instructions	AML Amendment Date
3	Piper	PA-11, PA-11S	A-691	CAR 4a	dated 8/26/08 or later FAA approved revision	dated March 10, 2008 and Document 9051-0171-002, Revision A, dated June 11, 2008, or later FAA accepted revision.	2/17/12
4	Piper	PA-12	A-780	CAR 3, amendment 3-8	R1710000-DL, Revision K, dated 10/27/11 or later FAA approved revision	Document 9051-0171-001, Revision D, dated February 16, 2012, or later FAA accepted revision.	2/17/12
5	Piper	PA-14	A-797	CAR 3, amendment 3-	R1710000-DL, Revision K, dated 10/27/11 or later FAA approved revision	Document 9051-0171-001, Revision D, dated February 16, 2012, or later FAA accepted revision.	2/17/12

Item	Aircraft Make	Aircraft Model	Original Type Certificate Number	Certification Basis for Alteration	Drawing List	Installation and Continued Airworthiness Instructions	AML Amendment Date
6	Piper	PA-15	A-800	CAR 3, amendment 3-8	Revision K, dated 10/27/11 or later FAA approved revision	Revision D, dated February 16, 2012, or later FAA accepted revision.	2/17/12
7	Piper	PA-16, PA-16S	1A1	CAR 3, amendment 3-8	R1710000-DL, Revision K, dated 10/27/11 or later FAA approved revision	Document 9051-0171-001, Revision D, dated February 16, 2012, or later FAA accepted revision.	2/17/12
8	Piper	PA-17	A-805	CAR 3, amendment 3-8	R1710000-DL, Revision K, dated 10/27/11 or later FAA approved revision	Document 9051-0171-001, Revision D, dated February 16, 2012, or later FAA accepted revision.	2/17/12

Item	Aircraft Make	Aircraft Model	Original Type Certificate Number	Certification Basis for Alteration	Drawing List	Installation and Continued Airworthiness Instructions	AML Amendment Date
9	Piper	PA-18, PA-18S, PA-18 "105" (Special), PA-18S "105" (Special), PA-18A, PA-18 "125" (Army L-21A), PA-18S "125", PA-18AS "125", PA-18 "135" (Army L-21B), PA-18A "135", PA-18S "135", PA-18AS "135", PA-18 "150", PA-18A "150", PA-18S "150", PA-18AS "150", PA-19 (Army L-18C), PA-19S	1A2	CAR 3, amendment 3-8	R1710000-DL, Revision K, dated 10/27/11 or later FAA approved revision	Document 9051-0171-001, Revision D, dated February 16, 2012, or later FAA accepted revision.	2/17/12
10	Piper	PA-20, PA-20S, PA-20 "115", PA-20S "115", PA-20 "135", PA-20S "135"	1A4	CAR 3, amendment 3-8	R1710000-DL, Revision K, dated 10/27/11 or later FAA approved revision	Document 9051-0171-001, Revision D, dated February 16, 2012, or later FAA accepted revision.	2/17/12
11	Piper	PA-22, PA-22-108, PA-22-135, PA-22S-135, PA-22-150, PA-22S-150, PA-22-160, PA-22S-160	1A6	CAR 3, amendment 3-8	R1710000-DL, Revision K, dated 10/27/11 or later FAA approved revision	Document 9051-0171-001, Revision D, dated February 16, 2012, or later FAA accepted revision.	2/17/12

Item	Aircraft Make	Aircraft Model	Original Type Certificate Number	Certification Basis for Alteration	Drawing List	Installation and Continued Airworthiness Instructions	AML Amendment Date
12	Luscombe	8A, 8B, 8C, 8D, 8E, 8F, T-8F	A-694	CAR-4a	RI171000-DL, Revision K, dated 10/27/11 or later FAA approved revision	later FAA accepted revision. Document 9051-0171-001, Revision D, dated February 16, 2012, or later FAA accepted revision.	2/17/12
13	Stinson	108-1, 108-1, 108-2, 108-3, 108-4	A-767	CAR-4a	RI171000-DL, Revision K, dated 10/27/11 or later FAA approved revision	Document 9051-0171-001, Revision D, dated February 16, 2012, or later FAA accepted revision.	2/17/12

FAA Approved: 
Acting Manager, Denver Aircraft Certification Office

Amended: February 28, 2012



U.S. Department
of Transportation
**Federal Aviation
Administration**

Transport Airplane Directorate
Aircraft Certification Service
Denver Aircraft Certification Office
26805 East 68th Avenue, Room 214
Denver, Colorado 80249

February 28, 2012

Mr. Gary Hanson
Rosen Sunvisor Systems LLC
86365 College View Road
Eugene, OR 97405

Dear Mr. Hanson:

This is in response to your November 17, 2011 request to amend Supplemental Type Certificate (STC) SA01251SE, for the purpose of installing a tubemount sunvisor system on additional aircraft defined on an approved model list (AML). Our review of the data submitted indicates that the requirements of Title 14 Code of Federal Regulations (14 CFR) have been satisfied. Amended STC SA01251SE, dated February 17, 2012, is enclosed.

In support of this amendment, this office also accepts Rosen's "Component Maintenance Manual with Illustrated Parts List", Document 9051-0171-001, Revision D, dated February 16, 2012, which includes instructions for continued airworthiness.

In accordance with Title 14 Code of Federal Regulations, 14 CFR 21.3, and as a recipient of this certificate, you are required to report any failure, malfunction, or defect, except as provided in 14 CFR 21.3(d), in any product or part manufactured by you or your contracted suppliers, and which you have determined has resulted or could result in any occurrences listed in 14 CFR 21.3(c). The report should be communicated initially by telephone to the Manager, Denver Aircraft Certification Office, telephone number (303) 342-1081, within 24 hours after it has been determined a failure has occurred and followed up with a written notice to the address shown above. FAA form 8010-4, Malfunction or Defect Report or other appropriate format is acceptable in transmitting the required details.

You, as the STC holder, are responsible for any design changes necessary to correct unsafe conditions as well as for submitting those design changes to this office for approval. This requirement is contained in 14 CFR 21.99. In addition, you are required to advise this office of any change of address or ownership.

Also, 14 CFR 21.50 requires that maintenance procedures for continued airworthiness, as applicable to this change in type design, be made available to the operator at the time the aircraft is returned to service.

By accepting this certificate, you acknowledge that you have read and understand your responsibilities as an STC holder.

If you plan to manufacture replacement or modification parts for sale in conformance with approved data listed on the Certificate, you are required to comply with 14 CFR 21.303. A Parts Manufacturer Approval (PMA) may be issued under the provisions of 14 CFR 21.303(d) when you submit a statement certifying you have established the fabrication inspection system as required by 14 CFR 21.303(h). The identification requirements for parts produced under a PMA are in 14 CFR 45.15. Your statement may be in letter form, with reference to STC SA01251SE, and should be addressed to the Federal Aviation Administration, Northwest Mountain Region, and Attention: Manager, Seattle Manufacturing Inspection District Office, at the address indicated below:

Federal Aviation Administration
Seattle Manufacturing Inspection District Office, ANM-108S
1801 Lind Ave. SW
Renton, WA 98057

If you have any questions, please contact Melissa Sandow at (303) 342-1084.

Sincerely,



Todd Dixon *per T. Dixon*
Manager, Denver
Aircraft Certification Office

Enclosure



Tube Mount NSA Sunvisor System

Date	Revision	Approved
10/2/15	M	GH

Drawing List
R1710000-DL

Kit R1710000-XXX

Doc. #9050-0171-001

1710200-001	R1710000				Drawing	Reference	Description	Rev
	001	014	034	041				
	1	1	1	1	1710000	R1710000	Complete System	G
1					1710200		Lens Assembly (Husky)	G
1	2	1			1710100-1		Mounting Assembly 3/4	G
			2		1710100-3		Mounting Assembly 5/8	G
			1					
				2	1710100-4		Mounting Assembly 7/8	G
			1					
1	2	1	2	2	1710102		Tube Mount – Swivel	D
			1	1				
1	2	1			1710103-1		Tube Mount – Bottom 3/4	J
			2		1710103-3		Tube Mount – Bottom 5/8	J
			1					
				2	1710103-4		Tube Mount – Bottom 7/8	J
			1					
1	2	1			1710104-1		Tube Mount – Top 3/4	H
			2		1710104-3		Tube Mount – Top 5/8	H
			1					
				2	1710104-4		Tube Mount – Top 7/8	H
			1					
1	2	1	2	2	1020100-001		NSA Mod Block Assembly	E
			1	1				
-	-	-	-	-	1020002-001		Modified 'A' Block	M
-	-	-	-	-	1020003-001		Modified 'B' Block	P
1	2	1	2	2	1010000-5		Complete Slide Assembly	E
			1	1				
-	-	-	-	-	1010001-5		Female Slide	L
-	-	-	-	-	1010002-3		Male Slide	L
	2				1710201		Lens	D
1	2	1	2	2	1010003		Lens Strip	D
			1	1				
1		1	2	2	1710202		Lens	C
			1	1				
1	1	1	1	1	9051-0171-001		Tube Mount CMM/IPC	D

Rosen®

SUNVISOR SYSTEMS

Part Number/Aircraft Model Application Table						
1710200	1710000					
-114	-011	-114	-034	-134	-041	-144
Husky A-1, A-1A, A-1B	Maule Bee Dee M-4, M-4, M-4C, M-4S, M-4T, M-4-180C, M- 4-180S, M-4- 180T, M-4-210, M-4-210C, M- 4-210S, M-4- 210T, M-4-220, M-4-220C, M- 4-220S, M-4- 220T, M-5- 180C, M-5-200, M-5-210C, M- 5-210TC, M-5- 220C, M-5- 235C, M-6-180, M-6-235, M-7- 235, MX-7-235, MX-7-180, MT- 7-235, M-8- 235, MX-7-160, MXT-7-160, MX-7-180A, MXT-7-180A, MX-7-420, MX-7-180B, M- 7-235B, M-7- 235A, M-7- 235C, MX-7- 180C, M-7-260, MT-7-260, M-7- 260C, M-7- 420AC, MX-7- 160C, MX-7- 180AC, M-7- 420A, MT-7- 420, MXT-7- 180 Luscombe 8, 8A, 8B, 8C, 8D, 8E, 8F, T- 8F	Piper PA-12, PA-14, PA-18, PA- 18S, PA-18 "105" (Special), PA-18S "105" (Special), PA-18A, PA-18 "125" (Army L- 21A), PA-18S "125", PA-18AS "125", PA-18 "135" (Army L- 21B), PA-18A "135", PA-18S "135", PA-18AS "135", PA-18 "150", PA-18A "150", PA-18S "150", PA-18AS "150", PA-19 (Army L-18C), PA-19S	Piper PA-20, PA-20S, PA-20 "115", PA-20S "115", PA-20 "135", PA-20S "135", PA-22, PA-22-108, PA-22-135, PA-22S-135, PA-22-150, PA-22S-150, PA-22-160, PA-22S-160	Piper PA-15, PA-16, PA-16S, PA-17	Stinson 108, 108-1, 108-2, 108-3, 108-4	Piper PA-11, PA-11S

Rosen®

SUNVISOR SYSTEMS

Installation Instructions for Husky Tube Mount Sunvisor System

This is an FAA STC'd installation requiring a logbook entry upon completion.

Doc: 9051-0171-002

Rev	Date	Approved
A	6-11-08	GH

Installing your Rosen NSA Sunvisor System is easily performed and should take approximately 1/2 to 1 hour.

*Please read these short instructions **COMPLETELY** before starting.*

Installation Hardware (included)

Qty: (1) 1/16 Allen Key for #6-32 Set Screw
 Qty: (1) 7/64 Allen Key for #6-32 Cap Screw
 Qty: (1) 9/64 Allen Key for #8-32 Cap Screw
 Qty: (1) 5/32 Allen Key for #10-32 Cap Screw
 Qty: (1) MS16995-27B #8-32 x 5/8 Socket Head Cap Screw
 Qty: (1) 8HCLW #8 High Collar Lock Washer
 Qty: (1) Lube Lube Packet

1. Loosen, but do not remove, Cap Screw #2 and Set Screw #1.
2. Remove 4 screws and lock washers from clamp assembly. Position top clamp half on tube approximately 5¼ inches from corner weld. Attach bottom clamp half using retained fasteners provided. Do not tighten at this point. (See images from a Husky installation on pages 2 & 3.)
3. Rotate clamp on tube and swivel post about axis 3 until swivel post is as near to vertical as possible. Note: This must be vertical in the flight attitude. Make snug, but do not tighten four (4) Clamp Screws and Cap Screw #2.
4. Use the provided Lube Packet to lightly lube the post of the clamp assembly
5. Place Visor Assembly on post and install using the #8-32 x 5/8 screw and #8 lock washer. Verify arm swings in a horizontal plane and does not interfere with any cabin feature.
6. Tighten four (4) clamp screws evenly. Tighten Screw #2 and Set Screw #1.
7. Tighten Screw # 5 until arm swings smoothly but not freely.
8. Installation is now complete.

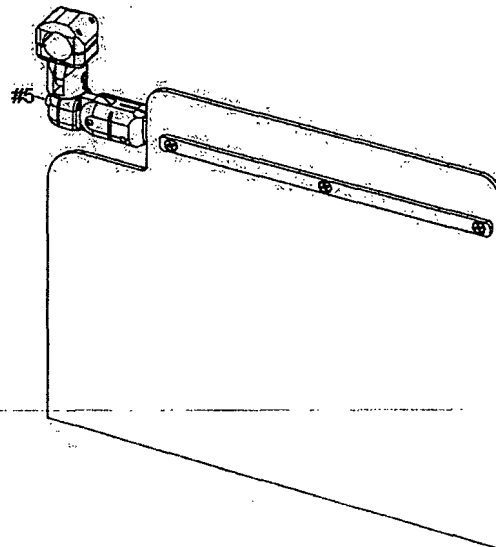
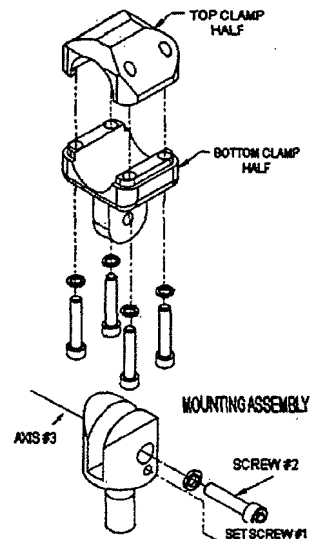




Image showing measurement from corner soft goods

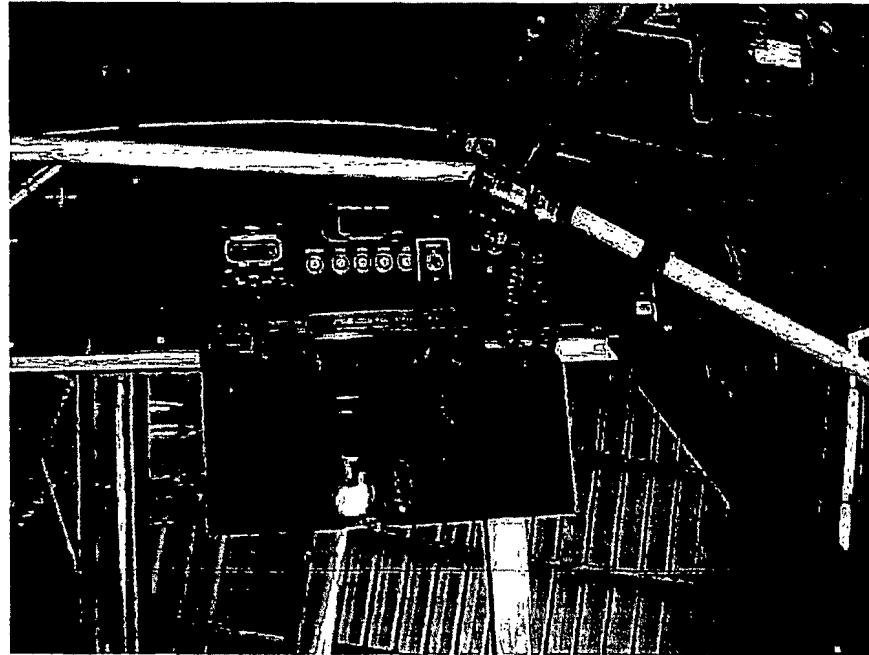


Image showing visor in side window protection position.

Continued Airworthiness Instructions:

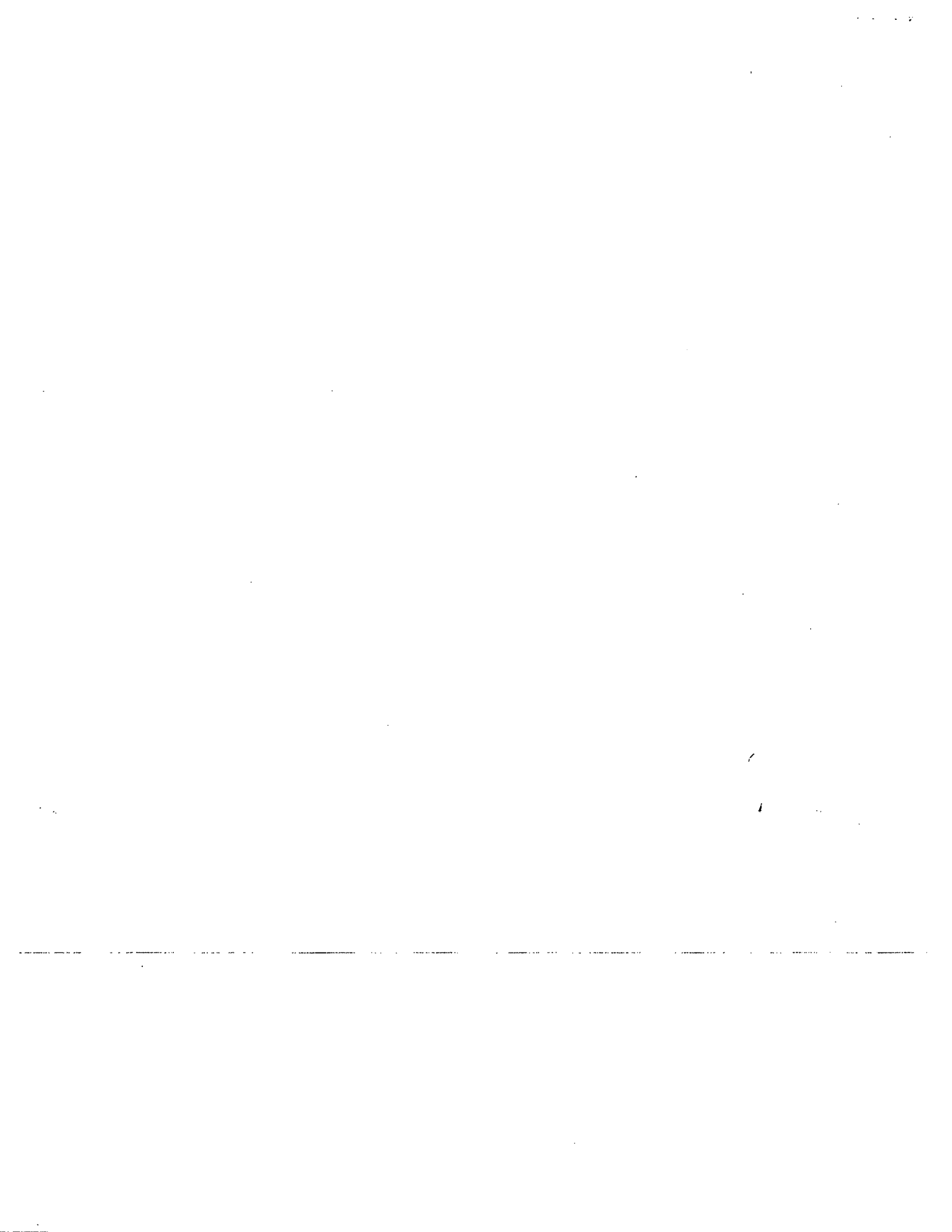
- **(On the ground only)**
 - Periodically clean the lenses a soft cloth using Rosen's Plastic Cleaner, Polisher and Protectant, or mild soap and water. Do not use abrasives on the lens.
 - Periodically adjust the pivot tensions on the visor assemblies.
- Updates to this continued airworthiness section are available on the Rosen Website. (www.rosenvisor.com)

The most up to date version of this document is available on the Rosen Website. (www.rosenvisor.com) We recommend that you periodically look to make sure you are using the most current version.

Airworthiness Limitations:

The Airworthiness Limitations Section is FAA approved and specifies maintenance required under §§43.16 and 91.403 of the Federal Aviation Regulations unless an alternative program has been FAA approved.

There are no airworthiness limitations associated with this installation.



DEPARTMENT OF TRANSPORTATION-FEDERAL AVIATION ADMINISTRATION

STANDARD AIRWORTHINESS CERTIFICATE

1 NATIONALITY AND REGISTRATION MARKS N67HY	2 MANUFACTURER AND MODEL Aviat Aircraft Inc. A-1B	3 AIRCRAFT SERIAL NUMBER 2065	4 CATEGORY Normal
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5 AUTHORITY AND BASIS FOR ISSUANCE

This airworthiness certificate is issued pursuant to 49 U.S.C. § 44704 and certifies that, as of the date of issuance, the aircraft to which issued has been inspected and found to conform to the type certificate therefor, to be in condition for safe operation, and has been shown to meet the requirements of the applicable comprehensive and detailed airworthiness code as provided by Annex 8 to the Convention on International Civil Aviation, except as noted herein.

Exceptions:

NONE

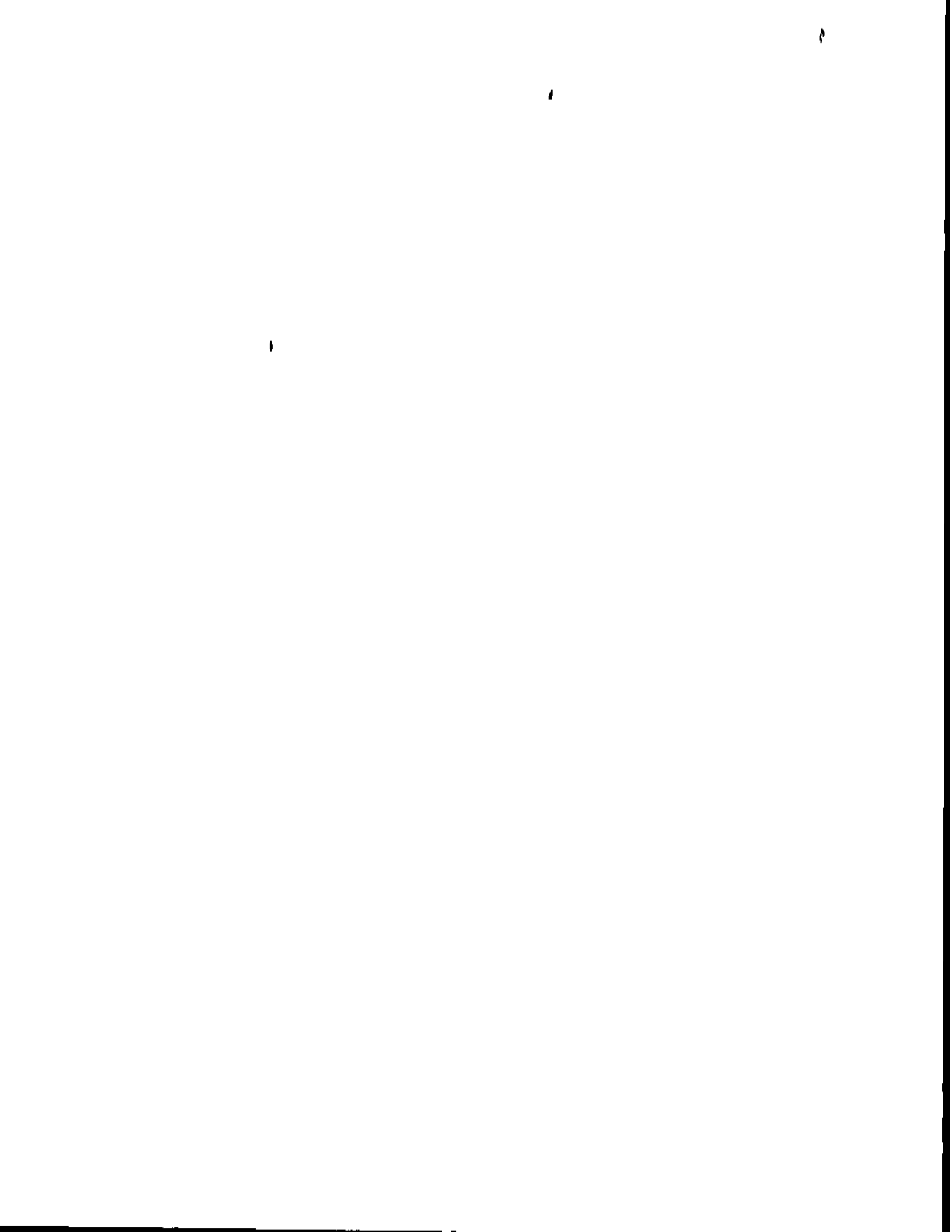
6 TERMS AND CONDITIONS

Unless sooner surrendered, suspended, revoked, or a termination date is otherwise established by the FAA, this airworthiness certificate is effective as long as the maintenance, preventative maintenance, and alterations are performed in accordance with Parts 21, 43, and 91 of the Federal Aviation Regulations, as appropriate, and the aircraft is registered in the United States.

DATE OF ISSUANCE R 03/07/2000	FAA REPRESENTATIVE Dennis E. Crawford 	DESIGNATION NUMBER GL-15
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Any alteration, reproduction, or misuse of this certificate may be punishable by a fine not exceeding \$1,000 or imprisonment not exceeding 3 years or both.

THIS CERTIFICATE MUST BE DISPLAYED IN THE AIRCRAFT IN ACCORDANCE WITH APPLICABLE FEDERAL AVIATION REGULATIONS.





WIPLINE FLOATS • SKIS • MODIFICATIONS • AIRCRAFT SALES
AVIONICS • INTERIOR • MAINTENANCE • PAINT REFINISHING

Federal Aviation Administration
Flight Standards District Office
6020 28th Ave So Room 201
Minneapolis, MN 55450
Dennis Crawford

5/6/15

Mr Crawford,

This letter is to request a replacement airworthiness certificate due to a lost certificate.

Original registration mark: N67HY
Aircraft Make: Aviat Aircraft Inc
Aircraft Model: A-1B
Aircraft serial number: 2065

Paul G Dauphinais
Chief Inspector
Wipaire Inc

11

Date 5/1/15

To whom it may concern

This letter authorizes a representative of Wipaire Inc to act as my agent for the purpose of applying for replacement airworthiness certificate due to lost certificate.

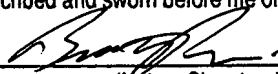
Aircraft Model Aviat Aircraft A-1B
Aircraft Serial 2065
Registration Mark N67HY

United States Department of the Interior
Office of Aviation Services
300 E. Mallard Dr., Suite 200
Boise, Idaho 83706-3991

Brian Green
Aviation Maintenance Specialist (Fleet):

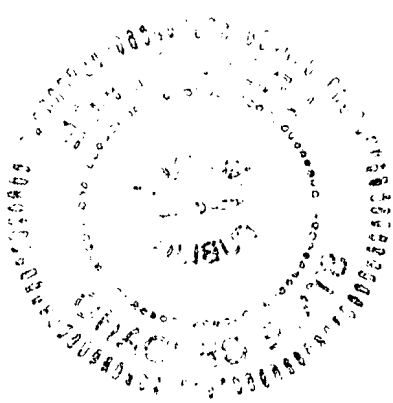


Notary Information

State of Idaho County of Ada
Subscribed and sworn before me on 05/04/2015
(Date)

(Notary Signature)



100



100



U.S. Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020
11/30/2007

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act 1958)

1. Aircraft	Nationality and Registration Mark N67HY	Serial No. 2065	
	Make AVIAT	Model A-1B	Series HUSKY
2. Owner	Name (As shown on registration certificate) U.S. DEPARTMENT OF INTERIOR		Address (As shown on registration certificate) 300 E. MALLARD DR. # 200
			City BOISE State ID Zip 83706 Country USA

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial Number
<input type="checkbox"/>	<input type="checkbox"/>	AIRFRAME	_____	(As described in Item 1 above)	_____
<input type="checkbox"/>	<input type="checkbox"/>	POWERPLANT			
<input type="checkbox"/>	<input type="checkbox"/>	PROPELLER			
<input checked="" type="checkbox"/>	<input type="checkbox"/>	APPLIANCE	Type FLOAT Manufacturer WIPLINE	2100S	21751 & 21752

6. Conformity Statement

A. Agency's Name and Address Name WIPAIRE INC Address 1700 HENRY AVE City SO ST PAUL State MN Zip 55075 Country USA	B. Kind of Agency <input type="checkbox"/> U.S. Certified Mechanic <input type="checkbox"/> Foreign Certified Mechanic <input checked="" type="checkbox"/> Certified Repair Station <input type="checkbox"/> Certified Maintenance Organization <input type="checkbox"/> Manufacturer C. Certificate No. RJWR390K
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D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B <input type="checkbox"/>	Signature/Date of Authorized Individual 4/4/2012 WESTON W BIGELOW
--	---

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

BY	FAA Fit Standards Inspector	Manufacturer	Maintenance Organization	Person Approved by Canadian Department of Transport
	FAA Designee	<input checked="" type="checkbox"/> Repair Station	Inspection Authorization	Other (Specify)

Certificate or Designation No. RJWR390K	Signature/Date of Authorized Individual 4/4/2012 WESTON W BIGELOW
---	---

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N67HY

4/4/2012

Nationality and Registration Mark

Date

REPAIRED FORWARD OUTBOARD SKIN OF RIGHT FLOAT BY SPLICING IN NEW SKIN FROM BULKHEAD #1 TO BULKHEAD #3 FROM GUNWALL TO CHINE USING SKIN P/N 21S01004-122 AS MATERIAL. USED DOUBLE ROW OF RIVETS AT SPLICE REFERENCING 2100 SERVICE MANUAL AND AC 43.13-1B FIG 4-16 AND CH 4-57. WEIGHT AND BALANCE CHANGE WAS NEGLIGIBLE. END*****

.....

Additional Sheets Are Attached



U.S. Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020
11/30/2007

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act 1958)

1. Aircraft	Nationality and Registration Mark N67HY	Serial No. 2065	
	Make AVIAT AIRCRAFT	Model A-1B	Series HUSKY
2. Owner	Name (As shown on registration certificate) U S DEPT OF THE INTERIOR		Address (As shown on registration certificate) 300 E MALLARD DR STE 200
			City BOISE State ID Zip 83706-3991 Country USA

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial Number
<input type="checkbox"/>	<input checked="" type="checkbox"/>	AIRFRAME	_____	(As described in Item 1 above)	_____
<input type="checkbox"/>	<input type="checkbox"/>	POWERPLANT			
<input type="checkbox"/>	<input type="checkbox"/>	PROPELLER			
<input type="checkbox"/>	<input type="checkbox"/>	APPLIANCE	Type		
			Manufacturer		

6. Conformity Statement

A. Agency's Name and Address Name WIPAIRE INC Address 1700 HENRY AVE City SO ST PAUL State MN Zip 55075 Country USA	B. Kind of Agency <input type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input checked="" type="checkbox"/> Certificated Repair Station <input type="checkbox"/> Certificated Maintenance Organization
	<input type="checkbox"/> Manufacturer C. Certificate No. RJWR390K

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B <input type="checkbox"/>	Signature/Date of Authorized Individual 9/3/09 PAUL G DAUPHINAIS <i>Paul G Dauphinais</i>
--	---

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

BY	FAA Fit Standards Inspector		Manufacturer	Maintenance Organization	Person Approved by Canadian Department of Transport
	FAA Designee	X	Repair Station	Inspection Authorization	Other (Specify)

Certificate or Designation No. RJWR390K	Signature/Date of Authorized Individual 9/3/09 PAUL G DAUPHINAIS <i>Paul G Dauphinais</i>
---	---

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N67HY

9/3/09

Nationality and Registration Mark

Date

INSTALLED WIPAIRE INC 2100S SEAPLANE FLOATS SN 21751 AND 21752 IAW STC SA00637CH DATED 7/23/08 AND INSTALLATION DRAWING 10D1-6009 REV F DATED 7/23/08. AIRCRAFT WEIGHED AND WEIGHT AND BALANCE REPORTS REVISED. END

Additional Sheets Are Attached



U.S. Department of
Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020
11/30/2007

Electronic Tracking Number

For FAA Use Only

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act 1958)

1. Aircraft	Nationality and Registration Mark N67HY	Serial No. 2065	
	Make Aviat	Model A-1B	Series
2. Owner	Name (As shown on registration certificate) US Department of Interior		Address (As shown on registration certificate) 300 East Mallard Dr. Suite 200
			City Boise State ID
			Zip 83706 Country US

3. For FAA Use Only

4. Type		5. Unit Identification			
Repair	Alteration	Unit	Make	Model	Serial Number
<input type="checkbox"/>	<input checked="" type="checkbox"/>	AIRFRAME	_____	(As described in Item 1 above)	_____
<input type="checkbox"/>	<input type="checkbox"/>	POWERPLANT			
<input type="checkbox"/>	<input type="checkbox"/>	PROPELLER			
<input type="checkbox"/>	<input type="checkbox"/>	APPLIANCE	Type		
			Manufacturer		

6. Conformity Statement

A. Agency's Name and Address Name Kevin J. Zubich Address 1414 16th St. East City Hibbing State MN Zip 55746 Country US	B. Kind of Agency <input checked="" type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input type="checkbox"/> Certificated Repair Station <input type="checkbox"/> Certificated Maintenance Organization <input type="checkbox"/> Manufacturer C. Certificate No. A&P2656709
---	---

D. I certify that the repair and/or alteration made to the unit(s) identified in item 5 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Extended range fuel per 14 CFR Part 43 App. B <input type="checkbox"/>	Signature/Date of Authorized Individual <i>Kevin J. Zubich</i> 4/14/08
--	---

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 5 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

BY	FAA Fit Standards Inspector	Manufacturer	Maintenance Organization	Person Approved by Canadian Department of Transport
	FAA Designee	Repair Station	<input checked="" type="checkbox"/> Inspection Authorization	Other (Specify)

Certificate or Designation No. 2656709IA	Signature/Date of Authorized Individual <i>Kevin J. Zubich</i> 4/14/08
--	---

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

N67HY

4/14/08

Nationality and Registration Mark

Date

Tube #49 Left was found damaged 3" aft of Tube 45. Repaired damage with welded splice repair I.A.W. AC 43.13-1B Change 1 Chapter 4 Section 5 Paragraph 4-95 Inner Sleeve Repairs. Primed and Restored Paint. Tube #49 Right was found Damaged 1" aft of Tube #96. Repaired damage with welded Splice Repair I.A.W. AC 43.13-1B Change 1 Chapter 4 Section 5 Paragraph 4-95 Inner Sleeve Repairs. Primed and Restored Paint. THE END

Additional Sheets Are Attached



US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification

217 66-15

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make Aviat	Model A-1B
	Serial No. 2065	Nationality and Registration Mark N67HY
2. Owner	Name (As shown on registration certificate) US Dept of Interior OAS	Address (As shown on registration certificate) 2350 W. Robinson Rd. Boise, ID. 83705-5355

3. For FAA Use Only

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	(As described in Item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address	B. Kind of Agency	C. Certificate No.
David L. Serena Ely Lake Aviation 3503 Ely Lake Dr. Eveleth, MN. 55734	<input checked="" type="checkbox"/> U.S. Certificated Mechanic	A&P472888877
	<input type="checkbox"/> Foreign Certificated Mechanic	
	<input type="checkbox"/> Certificated Repair Station	
	<input type="checkbox"/> Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 03/19/03	Signature of Authorized Individual <i>David L. Serena</i>
------------------	--

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

BY	FAA Fit. Standards Inspector	Manufacturer	X	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station		Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection 03/19/03		Certificate or Designation No. AP472888877IA	Signature of Authorized Individual <i>David L. Serena</i>		

NOTICE

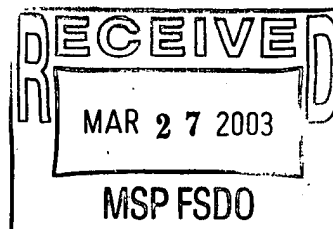
Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

1. Installed Prop-Guard Laminate as per Installation Instructions FP1001-CS1002 and STC SP00582 NY.

-----END-----



Additional Sheets Are Attached

United States of America
Department of Transportation -- Federal Aviation Administration

Supplemental Type Certificate

Number SP00582NY

This certificate issued to
Prop-Guard Aerospace, Inc.
922 Gorton Avenue
Cortland, NY 13045

certifies that the change in the type design for the following product with the limitations and conditions therefor as specified herein meets the airworthiness requirements of Part 14/35 of the Civil Air/Federal Aviation Regulations.

Original Product - Type Certificate Number: See Attached Approved

Models: Model List (AML)

Model: Model List (AML)

Description of Type Design Change:

Installation of Prop-Guard Polyurethane Protective Tape on the leading edge of metal propellers in accordance with the instructions contained in Prop-Guard Installation Instructions, Document No. FP1001, Rev. B dated June 3, 1999.

Limitations and Conditions:

1. Compatibility of this design change with previously approved modifications must be determined by the installer.
2. If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

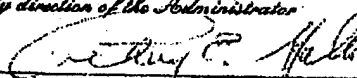
Date of application: April 05, 1996

Date of issuance: March 10, 1997

Date renewed:

Date amended: January 21, 1998; August 13, 1999

By direction of the Administrator


(Signature)

Vito A. Palera
Manager,
New York Aircraft Certification Office

(Title)



Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

FAA Form 8110-2(10-88)

Page 1 of 1

This certificate may be transferred in accordance with FAR 21.47.

FEDERAL AVIATION ADMINISTRATION - PARTS MANUFAC
ATTACHMENT 1 TO SUPPLEMENT NUMBE

ITEM	PROPELLER MAKE	PROPELLER MODEL (SERIES)	ORIGINAL TYPE CERTIFICATE	AML AMENDMENT DATE
48	Hamilton Standard Div.	61300T-1, 61300T-2	P139E	01/21/98
49	Hamilton Standard Div.	2D	P-333	01/21/98
50	Hamilton Standard Div.	2D	P-308	01/21/98
51	Hamilton Standard Div.	2E	377	01/21/98
52	Hamilton Standard Div.	1D	249	01/21/98
53	Hamilton Standard Div.	3E	348	01/21/98
54	Hamilton Standard Div.	12D	P-237	01/21/98
55	Hamilton Standard Div.	142F, 142F	P119E	01/21/98
56	Hamilton Standard Div.	22D	P-725	01/21/98
57	Hamilton Standard Div.	22D	P-719	01/21/98
58	Hamilton Standard Div.	22E	P-483	01/21/98
59	Hamilton Standard Div.	23F	764	01/21/98
60	Hamilton Standard Div.	23LF	P26EA	01/21/98
61	Hamilton Standard Div.	24D	P-724	01/21/98
62	Hamilton Standard Div.	24E	P-283	01/21/98
63	Hamilton Standard Div.	32D	P-349	01/21/98
64	Hamilton Standard Div.	32E	P-726	01/21/98
65	Hamilton Standard Div.	33LF	P26EA	01/21/98
66	Hamilton Standard Div.	34D	P-834	01/21/98
67	Hamilton Standard Div.	34E	P-831	01/21/98
68	Hamilton Standard Div.	43D	P-851	01/21/98
69	Hamilton Standard Div.	43E	P-871	01/21/98
70	Hamilton Standard Div.	43E	P-857	01/21/98
71	Hamilton Standard Div.	54E	P-805	01/21/98
72	Hamilton Standard Div.	63E	P136A	01/21/98

ITEM	PROPELLER MAKE	PROPELLER MODEL (SERIES)	ORIGINAL TYPE CERTIFICATE	AML AMENDMENT DATE
73	Hamilton Standard Div.	232	P-833	01/21/98
74	Hamilton Standard Div.	242	P-870	01/21/98
75	Hamilton Standard Div.	Aeroproducts A641FN	P-804	01/21/98
76	Hamilton Standard Div.	Aeroproducts A641FN	P-808	01/21/98
77	Hamilton Standard Div.	247F-1, 247F-2, 247F-3	P18D	01/21/98
78	Hamilton Standard Div.	8382F-1	P88D	01/21/98
79	Hamilton Standard Div.	145F-11, 145F-11E	P79E	01/21/98
80	Hartzell Propeller Inc.	HA-832	P-914	01/21/98
81	Hartzell Propeller Inc.	HA-12U	P-824	01/21/98
82	Hartzell Propeller Inc.	HC-A1X, HEC-A1X	P-908	01/21/98
83	Hartzell Propeller Inc.	HC-A1X, HEC-A1X	P-911	01/21/98
84	Hartzell Propeller Inc.	HC-812	P-298A	01/21/98
85	Hartzell Propeller Inc.	HC-834	P93E	01/21/98
86	Hartzell Propeller Inc.	HC-87T	P136A	01/21/98
87	Hartzell Propeller Inc.	HC-87Z	P-907	01/21/98
88	Hartzell Propeller Inc.	HC-87Y, HEC-87Y, HEC-87Y, HEC-87Y	P-920	01/21/98
89	Hartzell Propeller Inc.	HC-87Y, HEC-87Y, HEC-87Y	P68A	01/21/98
90	Hartzell Propeller Inc.	HC-87Y	P26EA	01/21/98
91	Hartzell Propeller Inc.	HC-87Y, HEC-87Y	P78A	01/21/98
92	Hartzell Propeller Inc.	HC-87Y, HEC-13Y, HEC-13Y	P136A	01/21/98
93	Hartzell Propeller Inc.	HC-12X	P-843	01/21/98
94	Hartzell Propeller Inc.	HC-13X	P-883	01/21/98
95	Hartzell Propeller Inc.	HC-422F	3P1	01/21/98
96	Hartzell Propeller Inc.	HC-53230	3P2	01/21/98
97	Hartzell Propeller Inc.	HC-82X	P-938	01/21/98

ITEM	PROPELLER MAKE	PROPELLER MODEL (SERIES)	ORIGINAL TYPE CERTIFICATE	AML AMENDMENT DATE
98	Hartzell Propeller Inc.	HC-81X	P-894	01/21/98
99	Hartzell Propeller Inc.	HC-92W, HEC-92W	P148A	01/21/98
100	Hartzell Propeller Inc.	HC-92Z, HEC-92Z	P-892	01/21/98
101	Hartzell Propeller Inc.	HC-93Z	P-891	01/21/98
102	Hartzell Propeller Inc.	HC-112Y	P-806	01/21/98
103	Hartzell Propeller Inc.	HC-11Y, HEC-11Y	P93E	01/21/98
104	Hartzell Propeller Inc.	HC-13Y, HEC-13Y	P148A	01/21/98
105	Hartzell Propeller Inc.	HC-82Y	P43E	01/21/98
106	Hartzell Propeller Inc.	HC-12Y	P43E	01/21/98
107	Hartzell Propeller Inc.	HC-87Y	P93E	01/21/98
108	Hartzell Propeller Inc.	HC-87Y	P51E	01/21/98
109	Hartzell Propeller Inc.	HC-84E, HEC-84E, HEC-84E, HEC-84E	P148E	01/21/98
110	Hartzell Propeller Inc.	HC-84T	P148E	01/21/98
111	Hartzell Propeller Inc.	HC-84Y	P53E	01/21/98
112	Hartzell Propeller Inc.	HC-84Y	P53E	01/21/98
113	Hartzell Propeller Inc.	HC-84Y	P53E	01/21/98
114	Hartzell Propeller Inc.	HC-87Y	P26EA	01/21/98
115	Hartzell Propeller Inc.	HC-87Y	P53E	01/21/98
116	Hartzell Propeller Inc.	HC-88A	P148E	01/21/98
117	Hartzell Propeller Inc.	HC-88M	P43E	01/21/98
118	Hartzell Propeller Inc.	HEC-87Y	P78A	01/21/98
119	Hartzell Propeller Inc.	HEC-13Y, HC-13Y	P78A	01/21/98
120	Hartzell Propeller Inc.	HC-87Y	P148E	01/21/98
121	Hartzell Propeller Inc.	HC-85	P89E	01/21/98
122	Hartzell Propeller Inc.	HC-85C	P90E	01/21/98

ITEM	PROPELLER MAKE	PROPELLER MODEL (SERIES)	ORIGINAL TYPE CERTIFICATE	AML AMENDMENT DATE
123	Hartzell Propeller Inc.	HC-87Y, HEC-87Y	P26EA	01/21/98
124	Hamilton Standard Div.	FD-312T	P-897	01/21/98
125	Hamilton Standard Div.	FD-406F	P-900	01/21/98
126	Hamilton Standard Div.	FD-406F	P-903	01/21/98
127	Hamilton Standard Div.	FD-312T	P-906	01/21/98
128	Hamilton Standard Div.	FD-322T	P-815	01/21/98
129	Hamilton Standard Div.	FD-476F	7P2	01/21/98
130	Hoffman Gable and Co. EO	137-B	7P2	01/21/98
131	Hoffman Gable and Co. EO	150-EMES	7P1	01/21/98
132	Hoffman Gable and Co. EO	HO288A, HO288B, HO288C	P37U	01/21/98
133	Hoffman Gable and Co. EO	HO-V 123	P38U	01/21/98
134	Hoffman Gable and Co. EO	HO-V82, HO-V82-B	P38U	01/21/98
135	Hoffman Gable and Co. EO	HO288A, HO288B, HO288C, HO288A, HO288B, HO288C, HO288A, HO288B, HO288C	P69E	01/21/98
136	Hoffman Gable and Co. EO	HO	P28E	01/21/98
137	Hoffman Gable and Co. EO	HO-V 132	TP58D	01/21/98
138	McCaughey Accessory Division	1A11D	838	01/21/98
139	McCaughey Accessory Division	C-1000	823	01/21/98
140	McCaughey Accessory Division	D-1055	815	01/21/98
141	McCaughey Accessory Division	MAC20 41D5925	821	01/21/98
142	McCaughey Accessory Division	2C36	835	01/21/98
143	McCaughey Accessory Division	2A31C, 2A31C	P-919	01/21/98
144	McCaughey Accessory Division	2A31C, 2A31C, 2A31C, 2A31C, 2A31C	P38A	01/21/98
145	McCaughey Accessory Division	2A31C, 2A31C, 2A31C, 2A31C	P-880	01/21/98
146	McCaughey Accessory Division	2A31C, 2A31C, 2A31C	P38A	01/21/98



U.S. Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved

OMB No. 2120-0020

For FAA Use Only

Office Identification
FAA-TAL-FSDO-03

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each violation (Section 901 of Federal Aviation Act of 1958).

1. Aircraft	Make <p style="text-align: center;">Aviat</p>	Model A-1B
	Serial No. 2065	Nationality and Registration Mark N67HY
2. Owner	Name (As shown on registration certificate) <p style="text-align: center;">Buena Vista Aviation, Inc.</p>	Address (As shown on registration certificate) <p style="text-align: center;">7911 Canal Street Anchorage, AK 99502</p>

3. For FAA Use Only

4. Unit Identification

5. Type

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	(As described in Item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency Name and Address Clifford D. Belleau 4340 POSTMARK DR. ANCHORAGE AK 99502	B. Kind of Agency <input checked="" type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input type="checkbox"/> Certified Repair Station <input type="checkbox"/> Manufacturer	C. Certificate No. 574-56-2293
---	--	--

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date <p style="font-size: 1.2em;">1-14-02</p>	Signature of Authorized Individual
--	--

7. Approval for/Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

BY	FAA Fit. Standards Inspector	Manufacturer	X	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station		Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection <p style="font-size: 1.2em;">1-14-02</p>		Certificate or Designation No. 574-56-2293	Signature of Authorized Individual 		

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished


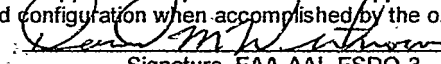
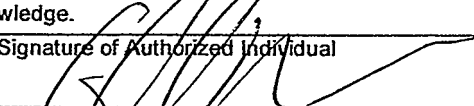
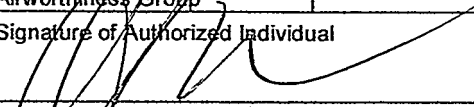
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

On N67HY:

1. Installed AvAlaska, Inc. shop fabricated oil cooler shutter kit #AvAk OC-1.
2. This installation duplicates the installation on N57HY. See FAA form 337 dated January 14, 2002.
3. An Airplane Flight Manual Supplement, document number N57HY, dated June 26, 2000 has been installed in the aircraft Approved Flight Manual.
4. Aircraft weight and balance and equipment list has been revised to reflect installation.
5. **Additional instructions for continued airworthiness beyond 14 CFR 43:** (1) Inspect shutter in the cooler shroud for condition at each annual or 100-hour inspection.

-----end of entry-----

N67HY 1-14-02

 US Department of Transportation Federal Aviation Administration	MAJOR REPAIR AND ALTERATION			Form Approved	
	(Airframe, Powerplant, Propeller, or Appliance)			OMB No. 2120-0020	
				For FAA Use Only	
Office Identification					
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each violation (Section 901 of Federal Aviation Act of 1958).					
1. Aircraft	Make	Aviat		Model A-1B	
	Serial No.	2057		Nationality and Registration Mark N57HY	
2. Owner	Name (As shown on registration certificate)			Address (As shown on registration certificate)	
	Buena Vista Aviation, Inc.			7911 Canal Street Anchorage, AK 99502	
3. For FAA Use Only					
The alteration identified herein complies with the applicable requirements and is approved for duplication on identical aircraft Make, model and altered configuration when accomplished by the original modifier.					
1/14/02 Date  Signature FAA-AAL-FSDO-3					
4. Unit Identification					5. Type
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	(As described in Item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				
6. Conformity Statement					
A. Agency Name and Address		B. Kind of Agency		C. Certificate No.	
Clifford D. Belleau 4340 POSTMARK DR. ANCHORAGE AK 99502		<input checked="" type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input type="checkbox"/> Certified Repair Station <input type="checkbox"/> Manufacturer		574-56-2293	
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.					
Date		Signature of Authorized Individual			
1-14-02					
7. Approval for Return to Service					
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED					
BY	FAA Fit. Standards Inspector	Manufacturer	<input checked="" type="checkbox"/>	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station		Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection		Certificate or Designation No.	Signature of Authorized Individual		
1-14-02		574-56-2293			

NOTICE.

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.):

On N57HY:

1. Installed AvAlaska, Inc. shop fabricated oil cooler shutter kit #AvAk OC-1.
2. The oil cooler push pull control is located on the left side of the instrument panel adjacent to the carburetor heat control. A Cessna P/N S1230-17 control actuates the shutter. The control is placarded "Oil cooler shutter push to open" and "Adjust oil cooler shutter to maintain correct oil temperature".
3. The control cable housing is secured along its run with tywraps and rubber stand offs to the fuselage tubular structure. The cable penetrates the firewall 2 1/2" to the left of the right hand upper engine mount bolt. The cable is protected by an MS35489-4 grommet and the hole is sealed with MIL-S-38249 firewall sealant. The forward end of the cable housing is secured to the oil cooler aft inboard upper bracket by one each MS21919WDG-2 Adel clamp, NAS74A3005P bushing, MS21045-3 nut and three AN960-10 washers. The forward end of the cable is wrapped 1.5 times around an AN52510R8 screw secured to the shutter with one each MS21042I3 nut and MS35650-302 nut.
4. The Aviat P/N 35391-502 oil cooler shroud was modified as follows: The original pop rivets used in its assembly were removed and four AN470AD4 rivets and six AN426AD4 rivets were used to secure the ends to the top plate. The inlet duct was removed from the top plate and a shop fabricated adapter plate of .032" thick 2024-T3 aluminum was riveted to the top using four AN470AD5 rivets in place of the four pop rivets originally installed. three .063" thick 2024-T3 spacers separate the inlet duct from the cooler shroud, and are installed with two AN426AD3 rivets and two AN426A5 rivets and one AN426A4 rivet. A MS21047L3K nut plate is secured to the shroud top plate in the forward inboard corner. A shop fabricated Shutter of .063" thick 2024-T3 aluminum is secured using one each AN52510R8 screw, and NAS75-3-003 bushing.
5. An Airplane Flight Manual Supplement, document number N57HY, dated June 26, 2000 has been installed in the aircraft Approved Flight Manual.
6. Aircraft weight and balance and equipment list has been revised to reflect installation.
7. **Additional instructions for continued airworthiness beyond 14 CFR 43:** (1) Inspect shutter in the cooler shroud for condition at each annual or 100-hour inspection.

-----end of entry-----

N67HY 1-14-02

AvAlaska, Inc.
4340 Postmark Dr.
Document Number N57HY

Airplane Flight Manual Supplement for
Sky International, Inc. (Aviat) Model Husky A-1B
N57HY; S/N 2057
With Oil cooler shutter Installed

AvAlaska, Inc.
FAA Approved
Airplane Flight Manual Supplement
For
Aviat A-1B

Serial Number: ~~2057~~ 2065
N Number: ~~N57HY~~ N67HY

This supplement must be attached to the FAA Approved Airplane Flight Manual dated January 28, 1998, or later FAA approved revision, when an AvAlaska, Inc. fabricated oil cooler shutter kit # AvAk OC-1 is installed in accordance with FAA Form 337 dated June 26, 2000. The information contained herein supplements or supersedes the information in the basic airplane flight manual only in those areas listed herein. For limitations, procedures and performance information not contained in this supplement, consult the basic airplane flight manual.

FAA Approved:

James P. Chudy

For Manager, Aircraft Certification Office
Federal Aviation Administration
Anchorage, Alaska

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N67HY 1-14-02

AvAlaska, Inc.
4340 Postmark Dr.
Document Number N57HY

Airplane Flight Manual Supplement for
Sky International, Inc. (Aviat) Model Husky A-1B
N57HY; S/N 2057
With Oil cooler shutter Installed

LOG OF REVISIONS

REVISION NUMBER	DATE	PAGES REVISED AND DESCRIPTION OF REVISIONS	APPROVAL SIGNATURE
-	6-26-00	ORIGINAL ISSUE	<i>James P. Chudy</i>

11
12

N67HY 1-14-02

AvAlaska, Inc.
4340 Postmark Dr.
Document Number N57HY

Airplane Flight Manual Supplement for
Sky International, Inc. (Aviat) Model Husky A-1B
N57HY; S/N 2057
With Oil cooler shutter Installed

SECTION I. Operating Limitations:

G. Markings and Placards: The following information must be displayed in the form of a Placard(s) next to the oil cooler shutter control knob:

Adjust oil cooler shutter to maintain correct oil temperature

Oil cooler shutter - PUSH TO OPEN

SECTION II. Normal Procedures:

D. STARTING ENGINE:

1A. Oil cooler shutterOPEN

E. BEFORE TAKE OFF:

5A. Oil cooler shutter AS REQUIRED

G. CRUISE:

4. Oil cooler shutter AS REQUIRED

H. BEFORE LANDING:

1A. Oil cooler shutter AS REQUIRED

SECTION III. Emergency Procedures:

B. ENGINE FAILURE DURING FLIGHT

3A. Oil cooler shutterOPEN

N67HY 1-14-02

AvAlaska, Inc.
4340 Postmark Dr.
Document Number N57HY

Airplane Flight Manual Supplement for
Sky International, Inc. (Aviat) Model Husky A-1B
N57HY; S/N 2057
With Oil cooler shutter Installed

- IV. Performance Information: No Change
- V. Loading Information: No Change



US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020
For FAA Use Only
Office Identification TAA-AL-FSDO-03

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each violation (Section 901 of Federal Aviation Act of 1958).

1. Aircraft	Make <p style="text-align: center;">Aviat</p>	Model A-1B
	Serial No. 2065	Nationality and Registration Mark N67HY
2. Owner	Name (As shown on registration certificate) <p style="text-align: center;">Buena Vista Aviation, Inc.</p>	Address (As shown on registration certificate) <p style="text-align: center;">7911 Canal Street Anchorage, AK 99502</p>

3. For FAA Use Only

4. Unit Identification

5. Type

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency Name and Address Clifford D. Belleau 4340 POSTMARK DR. ANCHORAGE AK 99502	B. Kind of Agency <input checked="" type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input type="checkbox"/> Certified Repair Station <input type="checkbox"/> Manufacturer	C. Certificate No. 574-56-2293
---	--	--

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date <p style="font-size: 1.2em;">1-14-02</p>	Signature of Authorized Individual
--	--

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

BY	FAA Fit. Standards Inspector	Manufacturer	x	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station		Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection <p style="font-size: 1.2em;">1-14-02</p>		Certificate or Designation No. 574-56-2293		Signature of Authorized Individual 	

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.):

On N67HY:

1. Installed F. A. Dodge Rear Seat Heater kit part number 3175-1, kit is approved for installation on Piper PA-18 series aircraft under STC SA02094AK. Installation on this Aviat aircraft is IAW the multiple field approval of N57HY. See FAA form 337 dated January 14, 2002.
2. An Airplane Flight Manual Supplement, document number N57HY-2, dated June 26, 2000 has been installed in the aircraft Approved Flight Manual.
3. Aircraft weight and balance and equipment list has been revised to reflect installation.
4. **Additional instructions for continued airworthiness beyond 14 CFR 43:** (1) Inspect flapper in the control box for condition at annual inspection, and that flapper is making a seal in the closed position. (2) Insure adequate clearance between carburetor heat shroud and cowling after any removal and reinstallation of shroud.

-----end of entry-----

N67HY 1-14-02

AvAlaska, Inc.
4340 Postmark Dr.
Document Number N57HY-2

Airplane Flight Manual Supplement for
Sky International, Inc. (Aviat) Model Husky A-1B
N57HY; S/N 2057
with Auxiliary Heater Installed

AvAlaska, Inc.
FAA Approved
Airplane Flight Manual Supplement
For
Aviat A-1B

Serial Number: ~~2057~~ 2065
N Number: ~~N57HY~~ N67HY

This supplement must be attached to the FAA Approved Airplane Flight Manual dated January 28, 1998, or later FAA approved revision, when F. A. Dodge Auxiliary Heater kit #3175 is installed in accordance with FAA Form 337 dated June 26, 2000. The information contained herein supplements or supersedes the information in the basic airplane flight manual only in those areas listed herein. For limitations, procedures and performance information not contained in this supplement, consult the basic airplane flight manual.

SECTION I. Operating Limitations:

G. Markings and Placards: The following information must be displayed in the form of a Placard(s) next to the oil cooler shutter control knob:

AUXILIARY HEAT -- PUSH TO CLOSE

SECTION III. Emergency Procedures:

D. FIRES*

3. FIRE IN FLIGHT (ENGINE)

d(a). AUXILIARY HEATCLOSE

IV. Performance Information: No Change

V. Loading Information: No Change

FAA Approved: James P. Chudy
Manager, Aircraft Certification Office
Federal Aviation Administration
Anchorage, Alaska

100
100



US Department
of Transportation
Federal Aviation
Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approval
ONB No. 2120-0020

For FAA Use Only

Office Identification

MSP FSDO G-45

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make Aviot Husky	Model A-1B
	Serial No. 2065	Nationality and Registration Mark N67HY
2. Owner	Name (As shown on registration certificate) US Dept of the Interior OAS	Address (As shown on registration certificate) 2350 W. Robinson Rd. Boise, ID. 83705-5355

3. For FAA Use Only
THE DATA IDENTIFIED HEREIN COMPLIES WITH APPLICABLE AIRWORTHINESS REQUIREMENTS AND IS APPROVED ONLY FOR THE ABOVE DESCRIBED AIRCRAFT SUBJECT TO CONFORMITY INSPECTION BY A PERSON AUTHORIZED IN FAR 43.7.

10/24/01 *Neil J. Amos*
DATE FAA INSPECTOR

4. Unit Identification

Unit	Make	Model	Serial No.	5. Type	
				Repair	Alteration
AIRFRAME	(As described in item 1 above)				XX
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

REJECTIVE
OCT 15 2001
MSP FSDO

6. Conformity Statement

A. Agency's Name and Address Northern Avionics 5825 Hoodoo Pt. Rd. Tower, MN. 55790	B. Kind of Agency		C. Certificate No. N5OR2020
		U.S. Certificated Mechanic	
		Foreign Certificated Mechanic	
	XX	Certificated Repair Station	
	Manufacturer		

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date October 02, 2001	Signature of Authorized Individual <i>Neil J. Amos</i> Neil Amos
--------------------------	--

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is **XX** APPROVED REJECTED

BY	FAA Fit. Standards Inspector	Manufacturer	Inspection Authorization	Other (Specify)
	FAA Designee	XX Repair Station	Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection October 02, 2001	Certificate or Designation No. N5OR2020	Signature of Authorized Individual <i>Neil J. Amos</i> Neil J. Amos
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NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.



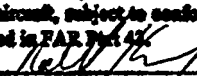
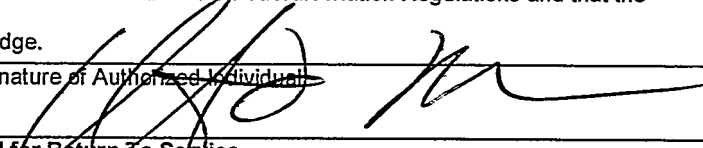
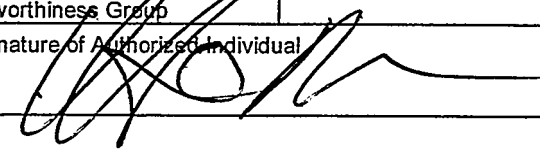
8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify aircraft nationality and registration mark and date work completed.)

1. Installed in the Item 1. (Aviot Husky A-1B) s/n 2065 N67HY the following:
 - a. King KT-76A Transponder s/n 126257
 - b. King KA-60 Antenna s/n 00020493
 - c. Ameri-King AK-350 Altitude Encoder s/n 41300
2. Appliances were installed under the following approved data:
 - a. King installation and service manual p/n 006-00143-0006
 - b. TSO Compliance C74b
 - c. Ameri-King installation manual and TSO-C88a
3. System was installed as a standalone ATCRBS system.
4. KT-76A was installed under the instrument panel below an existing antenna-switching unit at station 40
KA-60 antenna was installed on the forward belly cowl on the left of centerline of A/C.
AK-350 encoder was installed in the forward cabin area left side, to firewall. At station 30.
7. All work was completed IAW the following:
 - a. AC43-13-1B Chapter 11 sections 8-17 and Chapter 12 sections 2&3.
 - b. AC43-13-2A Par. 1 sections 4,6,9-11.
 - c. AC43-13-2A Par. 3 sections 36, 38a2, b1 & 3 and section 40 antenna installations.
8. Post installation ramp testing was completed successfully.
9. Instructions for Continued Airworthiness:
 1. Introduction: See section 1,
 2. Description: As above,
 3. Control: N/A,
 4. Servicing Information: There are no user-serviceable procedures for the system
 5. Maintenance Instructions: N/A,
 6. Trouble Shooting: Refer to KY76A operating handbook.,
 7. Removal and Replacement: Refer to manufacturers Installation manual for specific instructions on testing, removal and replacement.,
 8. Diagrams: N/A,
 9. Special Inspection Requirements: N/A,
 10. Application of protective treatments: N/A,
 11. Data: N/A,
 12. List of Special Tools: N/A
 13. For Commuter Aircraft: N/A,
 14. Recommended overhaul periods: N/A,
 15. Airworthiness Limitation Section: N/A,
 16. Revision: A letter will be submitted to the local FSDO with a copy of the revised Form 337 and ICA. Note: This ICA is now part of the aircraft's inspection and maintenance requirements.
10. Weight and Balance has been updated.

END

FAA

 US Department of Transportation Federal Aviation Administration	MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)				Form Approved	
					OMB No. 2120-0020	
					For FAA Use Only	
				Office Identification		
				AAL-FSDO-03 		
INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each violation (Section 901 of Federal Aviation Act of 1958).						
1. Aircraft	Make AVIAT			Model A-1B		
	Serial No. 2065			Nationality and Registration Mark N67HY		
2. Owner	Name (As shown on registration certificate) Buena Vista Aviation, Inc.			Address (As shown on registration certificate) 7911 Canal Street Anchorage, AK 99502		
	3. For FAA Use Only					
The alteration identified herein complies with the applicable airworthiness requirements and is approved for the above described aircraft, subject to conformity inspection by a person authorized in FAR Part 43. 06/27/00 						
Date		Signature		4. Unit Identification		5. Type
		FAA-AAL-FSDO-03				
Unit	Make	Model	Serial No.	Repair	Alteration	
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				X	
POWERPLANT						
PROPELLER						
APPLIANCE	Type					
	Manufacturer					
6. Conformity Statement						
A. Agency Name and Address			B. Kind of Agency		C. Certificate No.	
AVALASKA, INC. 4340 POSTMARK DR. ANCHORAGE AK 99502			<input type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input checked="" type="checkbox"/> Certified Repair Station <input type="checkbox"/> Manufacturer		V2AR414J	
D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.						
Date			Signature of Authorized Individual			
6-27-00						
7. Approval for Return To Service						
Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED						
BY	FAA Flt. Standards Inspector	Manufacturer	Inspection Authorization	Other (Specify)		
	FAA Designee	X Repair Station	Person Approved by Transport Canada Airworthiness Group			
Date of Approval or Rejection		Certificate or Designation No.		Signature of Authorized Individual		
6-27-00		V2AR414J				

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

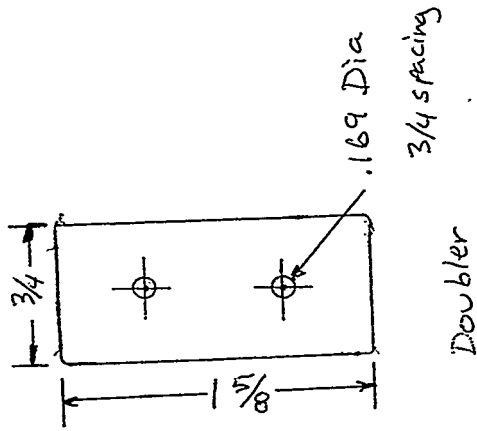
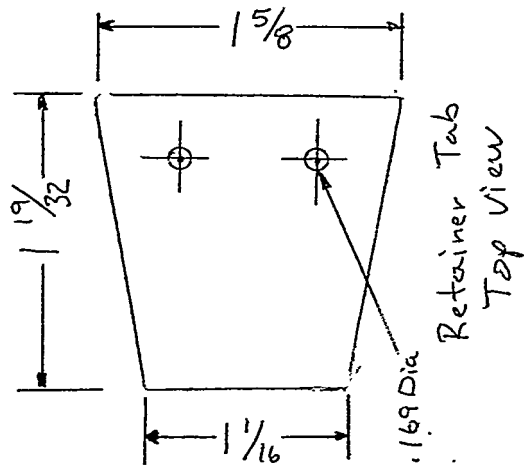
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.):

On N67HY:

1. Removed aft pilots control stick P/N 35240-002 and installed Airglass Engineering fabricated fiberglass stick control cover at aft pilots position.
2. Cover is secured at forward corners at station 104 by two each AN3-7A bolts, and NAS1149D0332J washers. The AN3 bolts are run into two existing nut plates that are used to secure the aft floorboard. The aft edges of the cover are secured by two shop fabricated .063" thick 2024-T3 aluminum tabs. The tabs are secured to the floorboard with two each MS27039-0812 screws, MS21044N08 nuts, and AN960-8 washers. A .063" thick 2024-T3 aluminum backing plate is used. See attached sketch of tabs.
3. Aircraft weight and balance and equipment list has been revised to reflect installation.
4. **Additional instructions for continued airworthiness beyond 14 CFR 43:** (1) Removal and installation of cover and control stick must be accomplished by an appropriately rated person IAW 14 CFR 43.3. (2) When the cover is removed the two forward retainer bolts must be replaced with AN526-1032R8 screws. When the control stick is installed the control stick retaining AN3-13A bolt, AN960-10 washer, and AN365-1032C nut must be reinstalled.

-----end of entry-----

Additional Sheets are Attached



Material
2024-T3
.063" Thick
Full size
N67HY
S/N 2065
Aviat A-1B
6-27-00




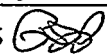
Retainer Tab
Side View

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125b

FAA

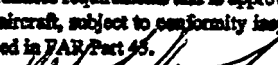
 US Department of Transportation Federal Aviation Administration	MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)	Form Approved OMB No. 2120-0020
		For FAA Use Only
		Office Identification <i>FAA-T300-03</i> 

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each violation (Section 901 of Federal Aviation Act of 1958).

1. Aircraft	Make AVIAT Serial No. 2065	Model A-1B Nationality and Registration Mark N67HY
2. Owner	Name (As shown on registration certificate) Buena Vista Aviation, Inc.	Address (As shown on registration certificate) 7911 Canal Street Anchorage, AK 99502

3. For FAA Use Only

The ~~unit~~ **alteration identified herein** complies with the applicable airworthiness requirements and is approved for the above described aircraft, subject to conformity inspection by a person authorized in FAR Part 43.

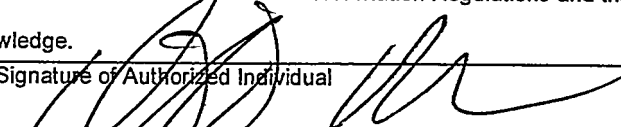
06/27/00 

Date	Signature / FAA-AAL-ESDO-03	4. Unit Identification	5. Type
Unit	Make	Model	Serial No.
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~		Repair
POWERPLANT			Alteration
PROPELLER			X
APPLIANCE	Type		
	Manufacturer		

6. Conformity Statement

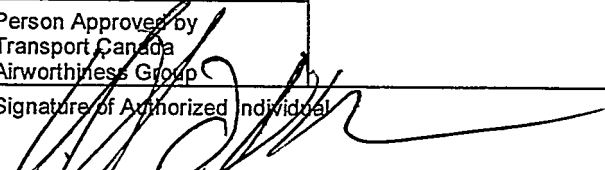
A. Agency Name and Address	B. Kind of Agency	C. Certificate No.
AVALASKA, INC. 4340 POSTMARK DR. ANCHORAGE AK 99502	<input type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input checked="" type="checkbox"/> Certified Repair Station <input type="checkbox"/> Manufacturer	V2AR414J

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date *6-27-00* Signature of Authorized Individual 

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

BY	FAA Fit. Standards Inspector		Manufacturer	Inspection Authorization	Other (Specify)
	FAA Designee	X	Repair Station	Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection			Certificate or Designation No.	Signature of Authorized Individual	
<i>6-27-00</i>			V2AR414J		

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.):

ON N67HY:

1. INSTALLED PITOT PRESSURE ACTIVATED HOURMETER SWITCH. SWITCH IS AVALASKA, INC. P/N AS9300-603. THIS SWITCH IS ELIGIBLE FOR INSTALLATION ON PIPER SINGLE ENGINE AIRCRAFT IAW STC SA02093AK.
2. UNIT IS INSTALLED IAW HOURMETER AIRSPEED SWITCH INSTALLATION INSTRUCTIONS FOR UNPRESSURIZED AIRPLANES, DATED DECEMBER 14, 1997.
3. SWITCH IS LOCATED AT F.S. 39. SWITCH IS SECURED TO FUSELAGE TUBING WITH TWO MS21919WDG12 ADEL CLAMPS AND KIT SUPPLIED HARDWARE.
4. AIRCRAFT WEIGHT AND BALANCE AND EQUIPMENT LIST REVISED TO REFLECT INSTALLATION.
5. NO ADDITIONAL INSTRUCTIONS FOR CONTINUED AIRWORTHINESS BEYOND 14 CFR 43.
END OF ENTRY

Additional Sheets are Attached

FAA



US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020
For FAA Use Only
Office Identification
AAL-FSDO-03

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each violation (Section 901 of Federal Aviation Act of 1958).

1. Aircraft	Make AVIAT	Model A-1B
	Serial No. 2065	Nationality and Registration Mark N67HY
2. Owner	Name (As shown on registration certificate) Buena Vista Aviation, Inc.	Address (As shown on registration certificate) 7911 Canal Street Anchorage, AK 99502

3. For FAA Use Only

The alteration identified herein complies with the applicable airworthiness requirements and is approved for the above described aircraft, subject to conformity inspection by a person authorized in FAR Part 43.
06/27/00

Date		Signature		4. Unit Identification		5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration		
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~						X
POWERPLANT							
PROPELLER							
APPLIANCE	Type						
	Manufacturer						

6. Conformity Statement

A. Agency Name and Address AVALASKA, INC. 4340 POSTMARK DR. ANCHORAGE AK 99502	B. Kind of Agency <input type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input checked="" type="checkbox"/> Certified Repair Station <input type="checkbox"/> Manufacturer	C. Certificate No. V2AR414J
---	---	--------------------------------

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 6-27-00	Signature of Authorized Individual
-----------------	------------------------------------

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

BY	FAA Flt. Standards Inspector	Manufacturer	Inspection Authorization	Other (Specify)
	FAA Designee	X Repair Station	Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection 6-27-00	Certificate or Designation No. V2AR414J	Signature of Authorized Individual		

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.):



On N67HY:

1. Installed 125 volt AC power plug, indicator light, battery blanket lead and associated wiring. The installation is connected to the previously installed Reiff Engine Preheat system. The system is used for pre-flight warming of the aircraft engine and battery. There is no electrical connection between the aircraft electrical system and this installation.
2. A flush mounted 15 amp. 125-volt AC power plug (Tanis P/N FMP) was installed on the right hand fuselage side panel (P/N 35358-502). A shop fabricated doubler and flush mounted cover plate are installed. The doubler is 2024-T3 aluminum of .032" thickness and is secured to side panel with thirteen MS20426 AD3-3.5 rivets. The cover plate is fabricated of 2024-T3 aluminum of .025" thickness, and is hinged with a 5/8" x 2 1/2" aluminum hinge on the forward side of the cover. A spring provides positive closure. The hinge is secured to the cover plate with three MS20470 AD3 rivets and to the side panel with three MS20470AD3-4 rivets. The power plug is secured with two each MS24693S272 screws and MS21044N3 nuts.
3. A 1/2" diameter 125 volt power indicator lamp assembly (Tanis P/N PLT) is installed forward of the power plug.
4. The system is wired with a 14 AWG, three conductor, low temperature type extension cord (SJOOW). One wire runs under the floorboard, through the firewall below the right hand rudder pedal box to the existing Reiff heater plug. The wire is chafe protected with a ms35489-12 grommet and the firewall sealed with CS-1900 firewall sealant. A second wire runs along the aircraft starter cable, aft to the battery area. The wire is terminated with a female plug end secured to fuselage tubing using one each MS21919WDG20 clamp, MS21919WDG10 clamp, MS35207-264 screw, MS21044N3 nut. The wires are secured along their runs to aircraft structure and wire bundles with tywraps.
5. Aircraft weight and balance and equipment list has been revised to reflect installation.
6. **No additional instructions for continued airworthiness beyond 14 CFR 43.**

-----end of entry-----

Additional Sheets are Attached

FAA

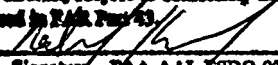
 US Department of Transportation Federal Aviation Administration	MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)	Form Approved
		OMB No. 2120-0020
		For FAA Use Only
		Office Identification <i>FAA-AAL/FSDO-03</i> 

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each violation (Section 901 of Federal Aviation Act of 1958).

1. Aircraft	Make AVIAT	Model A-1B
	Serial No. 2065	Nationality and Registration Mark N67HY
2. Owner	Name (As shown on registration certificate) Buena Vista Aviation, Inc.	Address (As shown on registration certificate) 7911 Canal Street Anchorage, AK 99502

3. For FAA Use Only

The ~~airframe~~ identified herein complies with the applicable airworthiness requirements and is approved for the above described aircraft, subject to conformity inspection by a person authorized in Part 43.

6/27/00 

Date <i>6/27/00</i>		Signature <i>FAA-AAL/FSDO-03</i>		4. Unit Identification		5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration		
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~						X
POWERPLANT							
PROPELLER							
APPLIANCE	Type						
	Manufacturer						

6. Conformity Statement

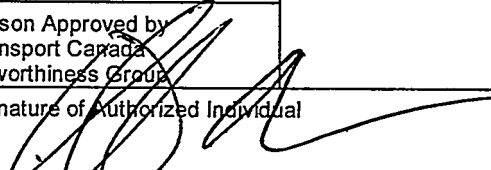
A. Agency Name and Address AVALASKA, INC. , 4340 POSTMARK DR. ANCHORAGE AK 99502	B. Kind of Agency <input type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input checked="" type="checkbox"/> Certified Repair Station <input type="checkbox"/> Manufacturer	C. Certificate No. V2AR414J
---	---	--------------------------------

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date <i>6-27-00</i>	Signature of Authorized Individual 
---------------------	---

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

BY	FAA Flt. Standards Inspector	Manufacturer	Inspection Authorization	Other (Specify)
	FAA Designee	<input checked="" type="checkbox"/> Repair Station	Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection <i>6-27-00</i>		Certificate or Designation No. V2AR414J	Signature of Authorized Individual 	

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.):

ON N67HY:

1. INSTALLED AVIAT AIRCRAFT, INC. AFT STOWAGE COMPARTMENT KIT P/N A-1-571 IAW SERVICE BULLITEN NO. 16 WITH THE FOLLOWING DEVIATIONS:
 - (A) RIGHT HAND FORWARD SIDE PANEL P/N 35655-006 WAS DISCARDED AND A MIRROR OF P/N 35577-501 ACCESS DOORFRAME WAS FABRICATED AND INSTALLED ON RH SIDE OF FUSELAGE. THE FRAME IS SECURED TO TOP LONGERON BY FOUR EACH AN742-12 CLAMPS, MS24693-S271 SCREWS, AND MS21044N3 NUTS. THE FRAME WAS SPLIT IN THREE PLACES TO ALLOW INSTALLATION UNDER FABRIC (UPPER, LOWER, AND FORWARD CHANNELS). THE FRAME WAS REASSEMBLED USING 2024-T3 .025" ALUMINUM SPLICE PLATES 4" LONG. FIVE MS210426 AD3 RIVETS ON EACH SIDE OF THE JOINT SECURE THE SPLICE PLATES. THE FORWARD AND LOWER SPLICES RECEIVED AN ADDITIONAL THREE RIVETS EACH SIDE OF THE JOINT. FORWARD SPLICE: THREE MS210426AD3 AND THREE AVIAT SSP-32 RIVETS. LOWER SPLICE SIX AVIAT SSP-32 RIVETS.
 - (B) A MIRROR OF P/N 35578-501 ACCESS DOOR WAS SHOP FABRICATED AND A P/N 35579-001 DRIP RAIL WAS INSTALLED ON THE RIGHT HAND SIDE OF FUSELAGE, AS ON LEFT HAND SIDE.
 - (C) RIGHT HAND AFT SIDE PANEL P/N 35655-008, WAS SHORTENED AND A 90 DEGREE BEND WAS MADE ON FRONT EDGE AND FIVE MS21069L06K NUT PLATES WERE INSTALLED.
 - (D) MIRRORS OF AFT DOOR SILL P/N 35655-018, BOTTOM TRIM ANGLE P/N 35655-019, AND FORWARD TRIM PANEL P/N 35655-020 WERE SHOP FABRICATED OF .032" 2024-T3 ALUMINUM AND INSTALLED ON RIGHT HAND SIDE OF FUSELAGE IN THE SAME MANNER AS THE LEFT HAND SIDE, WITH AN526C832R6 SCREWS AND MS21069L08K NUTPLATES REPLACING AN526C632 SCREWS AND BACN10FX61 NUTPLATES USED ON THE LEFT HAND SIDE ON THE FORWARD TRIM PANEL.
 - (E) THE SUPPORT ANGLE P/N 35655-034, WAS ELIMINATED.
 - (F) THE RIGHT HAND FORWARD TIE DOWN BRACKET P/N 35655-005 WAS TRIMMED TO CLEAR THE DOORSILL.
2. AIRCRAFT WEIGHT AND BALANCE AND EQUIPMENT LIST REVISED TO REFLECT INSTALLATION.
3. NO ADDITIONAL INSTRUCTIONS FOR CONTINUED AIRWORTHINESS BEYOND 14 CFR 43.
4. NO CHANGES TO FAA APPROVED AIRPLANE FLIGHT MANUAL SUPPLEMENT.

END OF ENTRY

Additional Sheets are Attached

FAA



US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020
For FAA Use Only

Office Identification
KAL-FSDO-03

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each violation (Section 901 of Federal Aviation Act of 1958).

1. Aircraft	Make AVIAT	Model A-1B
	Serial No. 2065	Nationality and Registration Mark N67HY
2. Owner	Name (As shown on registration certificate) Buena Vista Aviation, Inc.	Address (As shown on registration certificate) 7911 Canal Street Anchorage, AK 99502

3. For FAA Use Only

The ~~repair~~ fabrication identified herein complies with the applicable airworthiness requirements and is approved for the above described aircraft, subject to conformity inspection by a person authorized in FAR Part 43.

Date 06/27/00 Signature [Signature] FAA-KAL-FSDO-03

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency Name and Address AVALASKA, INC. 4340 POSTMARK DR. ANCHORAGE AK 99502	B. Kind of Agency <input type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input checked="" type="checkbox"/> Certified Repair Station <input type="checkbox"/> Manufacturer	C. Certificate No. V2AR414J
---	---	--------------------------------

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 6-27-00 Signature of Authorized Individual [Signature]

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

BY	FAA Flt. Standards Inspector	Manufacturer	Inspection Authorization	Other (Specify)
	FAA Designee	X Repair Station	Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection 6-27-00		Certificate or Designation No. V2AR414J	Signature of Authorized Individual [Signature]	

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.):

On N67HY:

1. Insulated cabin side, floor, and rear panels with Armstrong AP Armaflex, 1/4" thick sheet foam insulation. Insulation was glued to the inside of aircraft side panels using 3M Super 77 Spray Adhesive.
 2. Armaflex is certified to meet or exceed the requirements of 14 CFR 25.853. See attached certification.
 3. Aircraft weight and balance and equipment list has been revised to reflect installation.
 4. **Additional instructions for continued airworthiness beyond 14 CFR 43:** Inspect insulation for contamination with flammable products at each annual or 100-hour inspection. If contamination is found: remove or replace the affected section prior to further flight.
- end of entry-----

Additional Sheets are Attached



January 18, 2000

Howard
Office of Aircraft Services
FAX (907) 271-4787

Dear Howard:

This is to certify that Armstrong AP Armaflex Sheet meets the Airworthiness Standards: Transport Category Airplanes per Federal Aviation Regulation, Part 25 for Thermal and Acoustical Insulation and Insulation Covering, Air Ducting. The specific section is FAR 25.853.

The actual test values are as follows:

Sample	Flame time after flame removed (min/sec)	Drip Flame Time (sec)	Char length (In)
Requirements	15 seconds max	5 seconds max	8 Inch max
1	0:0	0 - No drip	3.10
2	0:0	0 - No drip	3.45
3	0:0	0 - No drip	3.80

The flame exposure time was 12 seconds.

If you have any additional questions, call me at (717) 396-4195. Thank you.

Sincerely,

Paul A. Hough
Manager, Technical Service
Insulation Products, N.A.

Aviat A-1B
S/N 2065
N67HY
6-27-00

folow

1
2
3
4

FAA



US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020
For FAA Use Only
Office Identification
AAI-FSDO-03

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each violation (Section 901 of Federal Aviation Act of 1958).

1. Aircraft	Make AVIAT	Model A-1B
	Serial No. 2065	Nationality and Registration Mark N67HY
2. Owner	Name (As shown on registration certificate) Buena Vista Aviation, Inc.	Address (As shown on registration certificate) 7911 Canal Street Anchorage, AK 99502

3. For FAA Use Only

The alteration identified herein complies with the applicable airworthiness requirements and is approved for the above described aircraft, subject to conformity inspection by a person authorized in FAR Part 43.
06/27/00

Date *06/27/00* Signature *[Signature]* FAA-AAI-FSDO-03

4. Unit Identification

5. Type

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency Name and Address AVALASKA, INC. 4340 POSTMARK DR. ANCHORAGE AK 99502	B. Kind of Agency <input type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input checked="" type="checkbox"/> Certified Repair Station <input type="checkbox"/> Manufacturer	C. Certificate No. V2AR414J
---	---	--------------------------------

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date <i>6-27-00</i>	Signature of Authorized Individual <i>[Signature]</i>
------------------------	--

7. Approval for Return to Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

BY	FAA Flt. Standards Inspector	Manufacturer	Inspection Authorization	Other (Specify)
	FAA Designee	<input checked="" type="checkbox"/> Repair Station	Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection <i>6-27-00</i>	Certificate or Designation No. V2AR414J	Signature of Authorized Individual <i>[Signature]</i>
---	--	--

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.):


ON N67HY:

1. INSTALLED HYDRAULIC SKI PUMP FITTINGS IN CABIN FLOOR AS FOLLOWS:
2. INSTALLED TWO EACH AN834-4D BULKHEAD TEE, AN960-716 WASHER, AN924-4D NUT IAW PAGE 8 OF AVIAT SERVICE BULLETIN NO. 15.
3. INSTALLED FOUR SHOP FABRICATED STRATOFLEX 111-4 HOSES ASSEMBLIES USING ONE 300-4S STRAIGHT FLARED FITTING AND ONE 649-4S 90 DEG. FLARED FITTING. LENGTHS ARE TWO EACH 16" LONG, ONE 11" LONG, ONE 10" LONG.
4. HOSES ARE PLUGGED WITH AN806-4D AND SECURED WITH TYWRAPS.
5. NO OTHER INSTALLATIONS WERE ACCOMPLISHED AT THIS TIME.
6. AIRCRAFT WEIGHT AND BALANCE AND EQUIPMENT LIST REVISED TO REFLECT INSTALLATION.
7. NO ADDITIONAL INSTRUCTIONS FOR CONTINUED AIRWORTHINESS BEYOND 14 CFR 43.

END OF ENTRY

Additional Sheets are Attached

FAA

 US Department of Transportation Federal Aviation Administration	MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)	Form Approved
		OMB No. 2120-0020
		For FAA Use Only
		Office Identification <i>AAL-FSDO-23628</i>

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each violation (Section 901 of Federal Aviation Act of 1958).

1. Aircraft	Make AVIAT	Model A-1B
	Serial No. 2065	Nationality and Registration Mark N67HY
2. Owner	Name (As shown on registration certificate) Buena Vista Aviation, Inc.	Address (As shown on registration certificate) 7911 Canal Street Anchorage, AK 99502

3. For FAA Use Only

The alteration identified herein complies with the applicable airworthiness requirements and is approved for the above described aircraft, subject to conformity inspection by a person authorized in FAR Part 43.

Date *6/27/00* Signature *[Signature]* FAA-AAL-FSDO-03

Unit	Make	Model	Serial No.	5. Type	
				Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency Name and Address AVALASKA, INC. 4340 POSTMARK DR. ANCHORAGE AK 99502	B. Kind of Agency	C. Certificate No. V2AR414J
	<input type="checkbox"/> U.S. Certificated Mechanic	
	<input type="checkbox"/> Foreign Certificated Mechanic	
	<input checked="" type="checkbox"/> Certified Repair Station	
	Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date *6-27-00* Signature of Authorized Individual *[Signature]*

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

BY	FAA Flt. Standards Inspector	Manufacturer	Inspection Authorization	Other (Specify)
	FAA Designee	<input checked="" type="checkbox"/> Repair Station	Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection <i>6-27-00</i>		Certificate or Designation No. V2AR414J	Signature of Authorized Individual <i>[Signature]</i>	

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.):

On N67HY:

1. Removed standard aft seat lap belt and inertia reel 5 point shoulder harness assembly.
2. Installed Aircraft Belts Inc. lap belt and inertia reel 5 point shoulder harness assembly P/N SBCB17089311-BOAA-601 meeting TSO c114.
3. Inertia reel was secured to airframe using original attach hardware and an NAS43DD3-27 bushing reamed out to 1/4" ID.
4. Aircraft weight and balance and equipment list has been revised to reflect installation.
5. **Additional instructions for continued airworthiness beyond 14 CFR 43: inspect belts at each annual or 100 hour inspection IAW Aircraft Belts, Inc. instructions for Continued Airworthiness, dated 10/30/97 or later revision.**

-----end of entry-----

Additional Sheets are Attached

IR 10/30/97

AIRCRAFT BELTS, INC. MAINTENANCE INSTRUCTIONS FOR CONTINUED AIRWORTHINESS

IF ANY OF THE FOLLOWING CONDITIONS ARE CHECKED 'YES' YOU SHOULD RETURN YOUR BELTS FOR REPAIR IMMEDIATELY.

WEBBING

YES NO

___ ___
___ ___
___ ___

(FRAYED) WHEN INDIVIDUAL STRANDS OF YARN BEGIN TO UNRAVEL
(TORN) WHEN STRANDS BECOME SEVERED
(FADING) WHEN UNCOVERED AREAS OF WEBBING DIFFER FROM COVERED AREAS BY 2 SHADES. USE GRAY SCALE FOR EVALUATING STAINING.

TO OBTAIN A GRAY SCALE, WRITE TO:

**AMERICAN ASSOCIATION OF TEXTILE
CHEMISTS AND COLORISTS
P.O. BOX 12215
RESEARCH TRIANGLE PARK, N.C. 27709 USA**

WEBBING SHOULD BE CLEANED WITH A MILD LAUNDRY DETERGENT ONLY. USE COLD WATER AND DO NOT HEAT TO DRY, AIR DRY ONLY.

HARDWARE

YES NO

___ ___
___ ___

(HOOK ENDFITTINGS OR BOLT ON ENDFITTINGS) BENT, INOPERABLE
SPRINGS, CLIPS, OR LATCHES; BOLT HOLES WORN OPEN; CORROSION OR RUST
(BUCKLE & CONNECTOR) INOPERATIONAL; BUCKLE LID OR CONNECTOR
BENT; DIFFICULT TO ADJUST; BUCKLE SHOWS DEFORMITY; RUST OR CORROSION

STITCHES

YES NO

___ ___
___ ___
___ ___

BROKEN OR MISSING STITCHES
SEVERE FADING OR DISCOLORING (MORE THAN TWO SHADES)
INCONSISTENT STITCH PATTERN (MAY INDICATE AN ALTERING OF
ASSEMBLY BY UNAUTHORIZED, UNQUALIFIED PERSONS, IT IS ILLEGAL AND UN-SAFE)

TSO TAGS

YES NO

___ ___
___ ___

TSO TAGS ARE MISSING OR ILLEGIBLE
DATE ON TSO TAG AGING THE BELTS AT 10 YEARS OR MORE (MAYBE
UNSAFE)

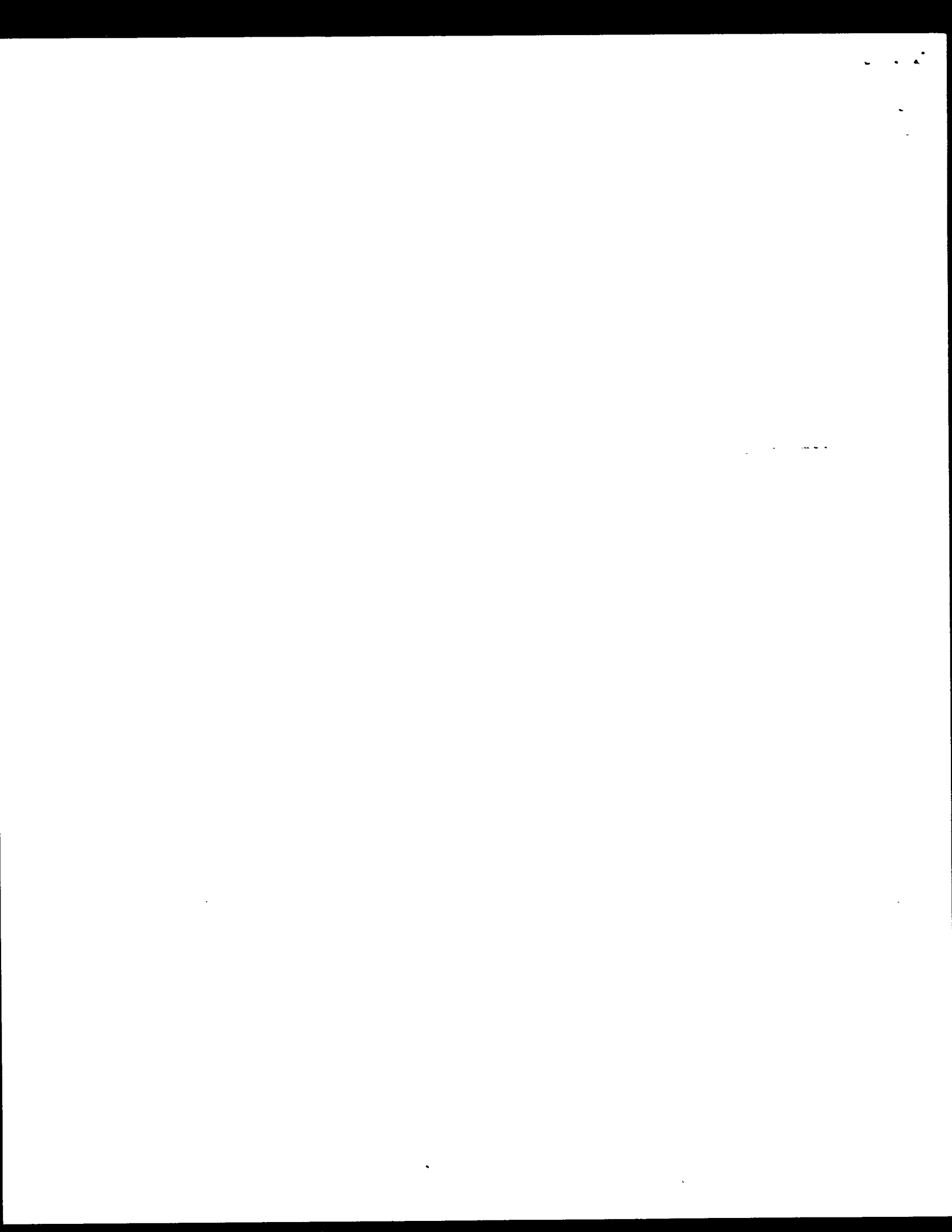
THE CONDITIONS AND TEST REQUIRED FOR TSO APPROVAL OF THIS ARTICLE ARE MINIMUM PERFORMANCE STANDARDS. IT IS THE RESPONSIBILITY OF THOSE DESIRING TO INSTALL THE ARTICLE EITHER ON OR WITHIN A SPECIFIC TYPE OR CLASS OF AIRCRAFT TO DETERMINE THAT THE AIRCRAFT INSTALLATION CONDITIONS ARE WITHIN THE TSO STANDARDS. IF NOT WITHIN THE TSO STANDARDS, THE ARTICLE MAY BE INSTALLED ONLY IF FURTHER EVALUATION BY THE APPLICANT DOCUMENTS AN ACCEPTABLE INSTALLATION AND IS APPROVED BY THE ADMINISTRATOR.

Aviat A-1B

SN 2065

N67HY

6-27-00



FAA



US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020
For FAA Use Only
Office Identification
AAI-150003

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each violation (Section 901 of Federal Aviation Act of 1958).

1. Aircraft	Make AVIAT	Model A-1B
	Serial No. 2065	Nationality and Registration Mark N67HY
2. Owner	Name (As shown on registration certificate) Buena Vista Aviation, Inc.	Address (As shown on registration certificate) 7911 Canal Street Anchorage, AK 99502

3. For FAA Use Only

The alteration identified herein complies with the applicable airworthiness requirements and is approved for the above described aircraft, subject to conformity inspection by a person authorized in FAR Part 43.

Date: 06/27/00 Signature: [Signature] FAA-AAI-FSDO-03

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency Name and Address AVALASKA, INC. 4340 POSTMARK DR. ANCHORAGE AK 99502	B. Kind of Agency		C. Certificate No. V2AR414J
	<input type="checkbox"/>	U.S. Certificated Mechanic	
	<input type="checkbox"/>	Foreign Certificated Mechanic	
	<input checked="" type="checkbox"/>	Certified Repair Station	
Manufacturer			

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date: 6-27-00 Signature of Authorized Individual: [Signature]

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

BY	FAA Flt. Standards Inspector	Manufacturer	Inspection Authorization	Other (Specify)
	FAA Designee	X Repair Station	Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection 6-27-00		Certificate or Designation No. V2AR414J	Signature of Authorized Individual [Signature]	

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.):

ON N67HY:

1. INSTALLED FUELING STEPS ON LANDING GEAR LEG FORWARD TUBES.
 2. STEPS ARE F.A. DODGE P/N 3190 STEPS THAT HAVE BEEN MODIFIED BY REMOVING THE CLAMPS AND WELDING ONE EACH MS27405T10P CLAMP ONTO EACH STEP. THIS WAS REQUIRED AS THE AVIAT GEAR LEG IS 1 1/4" DIAMETER. THE PIPER GEAR LEG IS 1 1/2" DIAMETER.
 3. 2 3/8" X 1 1/8" HOLES WERE CUT IN THE GEAR LEG FAIRINGS ON THE IN AND OUTBOARD SKINS OVER THE FORWARD GEAR LEG TUBES TO ALLOW THE STEPS TO BE INSTALLED. THREE SHOP FABRICATED COVER PLATES OF 2024-H3 .025" THICK ALUMINUM WERE INSTALLED TO SEAL THE OPENINGS. THE COVER PLATES ARE SECURED WITH FOUR EACH 4R X 1/4" TYPE A SHEET METAL SCREWS.
 4. AIRCRAFT WEIGHT AND BALANCE AND EQUIPMENT LIST REFLECT INSTALLATION.
 5. NO ADDITIONAL INSTRUCTIONS FOR CONTINUED AIRWORTHINESS BEYOND 14 CFR 43.
- END OF ENTRY

Additional Sheets are Attached

FAA



US Department of Transportation
Federal Aviation Administration

**MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)**

Form Approved
OMB No. 2120-0020
For FAA Use Only
Office Identification
FAA-AAL-FSDO-03 *RTK*

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each violation (Section 901 of Federal Aviation Act of 1958).

1. Aircraft	Make Aviat	Model A-1B
	Serial No. 2065	Nationality and Registration Mark N67HY
2. Owner	Name (As shown on registration certificate) Buena Vista Aviation, Inc.	Address (As shown on registration certificate) 7911 Canal Street Anchorage, AK 99502

3. For FAA Use Only

4. Unit Identification

5. Type

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	----- (As described in Item 1 above) -----				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency Name and Address Clifford D. Belleau 4340 POSTMARK DR. ANCHORAGE AK 99502	B. Kind of Agency <input checked="" type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input type="checkbox"/> Certified Repair Station <input type="checkbox"/> Manufacturer	C. Certificate No. 574-56-2293
--	---	-----------------------------------

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date <i>6-27-00</i>	Signature of Authorized Individual <i>[Signature]</i>
------------------------	--

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

BY	FAA Flt. Standards Inspector	Manufacturer	<input checked="" type="checkbox"/>	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station		Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection <i>6-27-00</i>		Certificate or Designation No. 574-56-2293	Signature of Authorized Individual <i>[Signature]</i>		

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.):

On N67HY:

1. Installed AvAlaska, Inc. shop fabricated oil cooler shutter kit #AvAk OC-1.
2. This installation duplicates the installation on N57HY. See FAA form 337 dated June 26, 2000.
3. An Airplane Flight Manual Supplement, document number N57HY, dated June 26, 2000 has been installed in the aircraft Approved Flight Manual.
4. Aircraft weight and balance and equipment list has been revised to reflect installation.
5. **Additional instructions for continued airworthiness beyond 14 CFR 43:** (1) Inspect shutter in the cooler shroud for condition at each annual or 100-hour inspection.

-----end of entry-----

Additional Sheets are Attached

AvAlaska, Inc.
4340 Postmark Dr.
Document Number N57HY

Airplane Flight Manual Supplement for
Sky International, Inc. (Aviat) Model Husky A-1B
N57HY; S/N 2057
With Oil cooler shutter Installed

AvAlaska, Inc.
FAA Approved
Airplane Flight Manual Supplement
For
Aviat A-1B

Serial Number: ~~2057~~ 2065
N Number: ~~N57HY~~ N67HY

This supplement must be attached to the FAA Approved Airplane Flight Manual dated January 28, 1998, or later FAA approved revision, when an AvAlaska, Inc. fabricated oil cooler shutter kit # AvAk OC-1 is installed in accordance with FAA Form 337 dated June 26, 2000. The information contained herein supplements or supersedes the information in the basic airplane flight manual only in those areas listed herein. For limitations, procedures and performance information not contained in this supplement, consult the basic airplane flight manual.

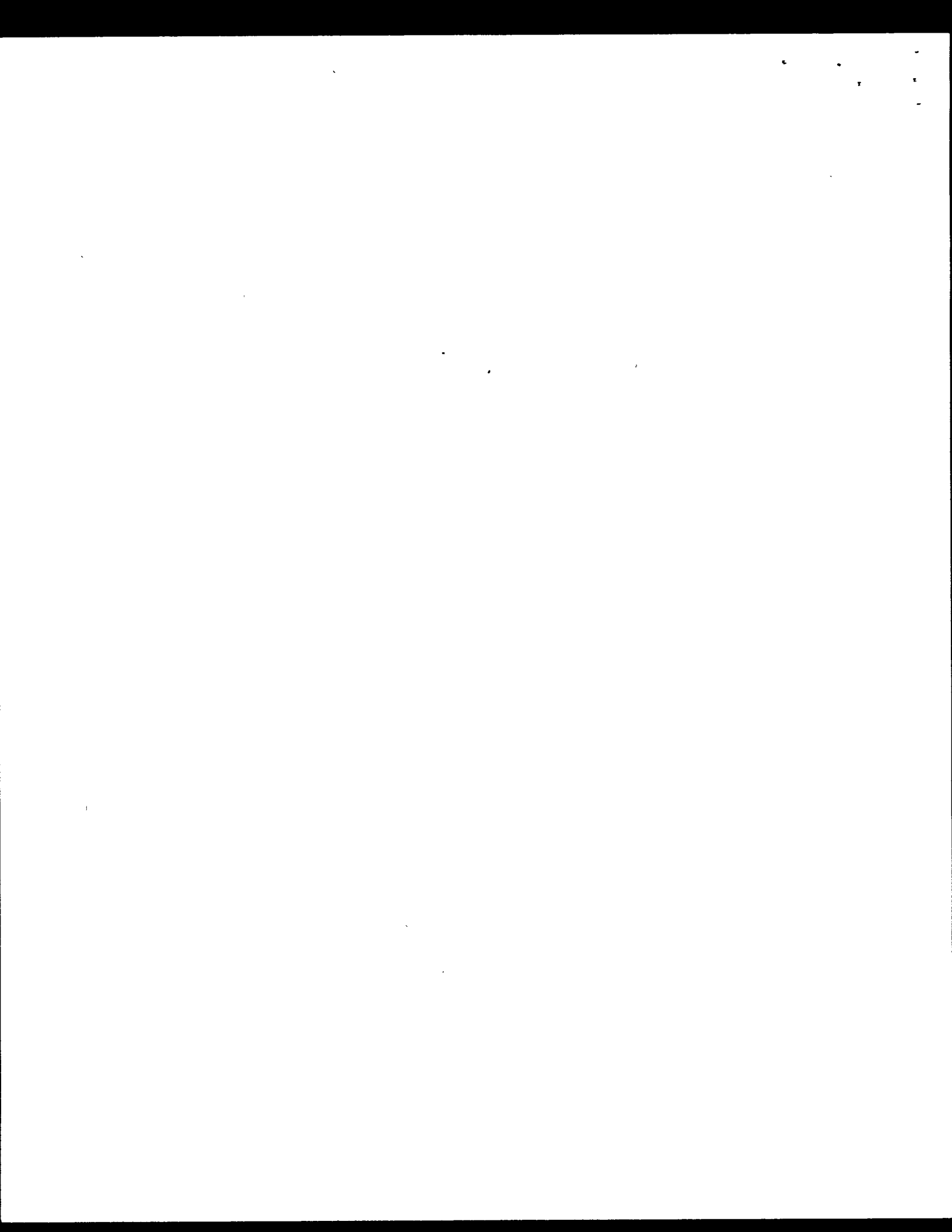
FAA Approved:



Manager, Aircraft Certification Office
Federal Aviation Administration
Anchorage, Alaska

DATE: June 26, 2000

Page 1 of 4

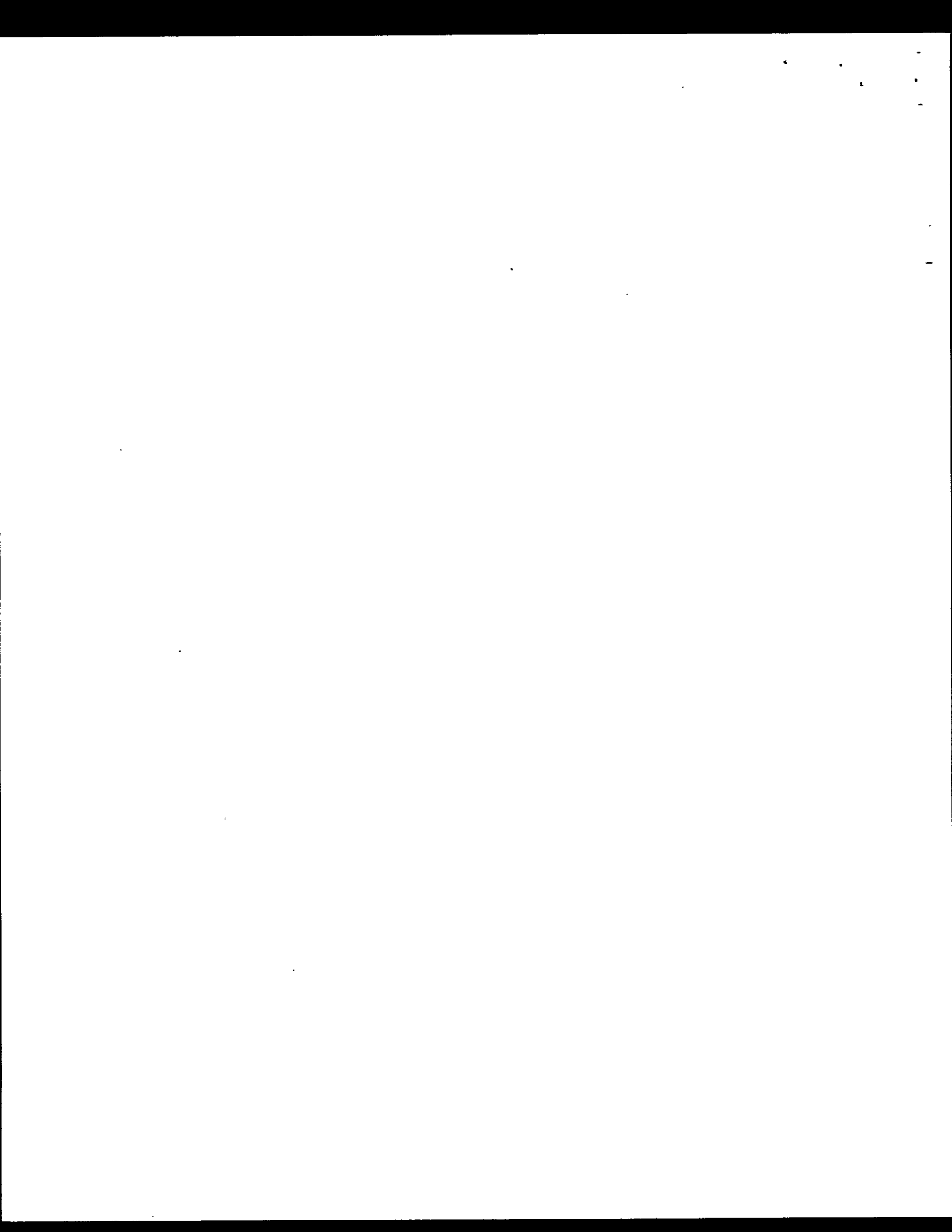


AvAlaska, Inc.
4340 Postmark Dr.
Document Number N57HY

Airplane Flight Manual Supplement for
Sky International, Inc. (Aviat) Model Husky A-1B
N57HY; S/N 2057
With Oil cooler shutter Installed

LOG OF REVISIONS

REVISION NUMBER	DATE	PAGES REVISED AND DESCRIPTION OF REVISIONS	APPROVAL SIGNATURE
-	6-26-00	ORIGINAL ISSUE	<i>James P. Chudy</i>



SECTION I. Operating Limitations:

G. Markings and Placards: The following information must be displayed in the form of a Placard(s) next to the oil cooler shutter control knob:

Adjust oil cooler shutter to maintain correct oil temperature

Oil cooler shutter - PUSH TO OPEN

SECTION II. Normal Procedures:

D. STARTING ENGINE:

1A. Oil cooler shutterOPEN

E. BEFORE TAKE OFF:

5A. Oil cooler shutter AS REQUIRED

G. CRUISE:

4. Oil cooler shutter AS REQUIRED

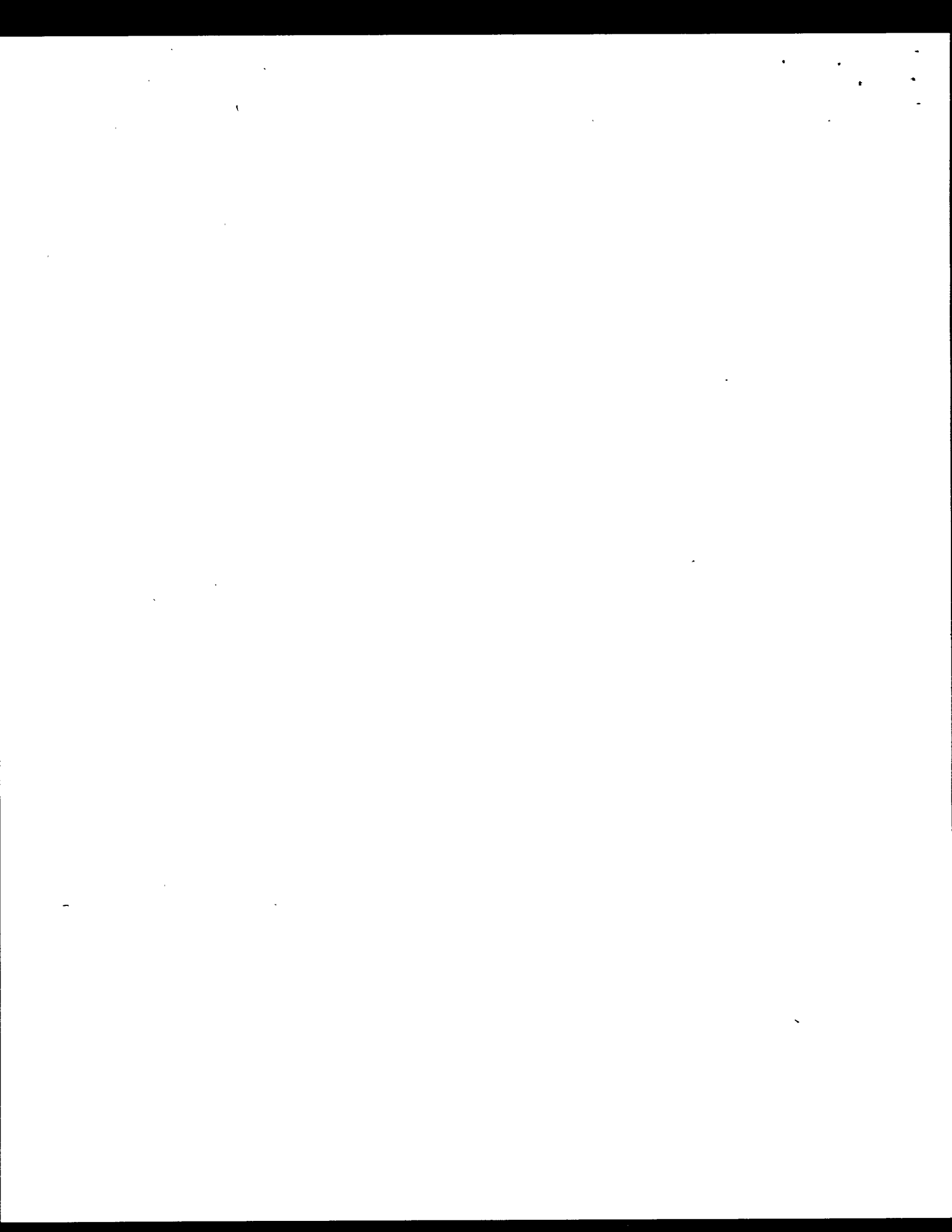
H. BEFORE LANDING:

1A. Oil cooler shutter AS REQUIRED

SECTION III. Emergency Procedures:

B. ENGINE FAILURE DURING FLIGHT

3A. Oil cooler shutterOPEN

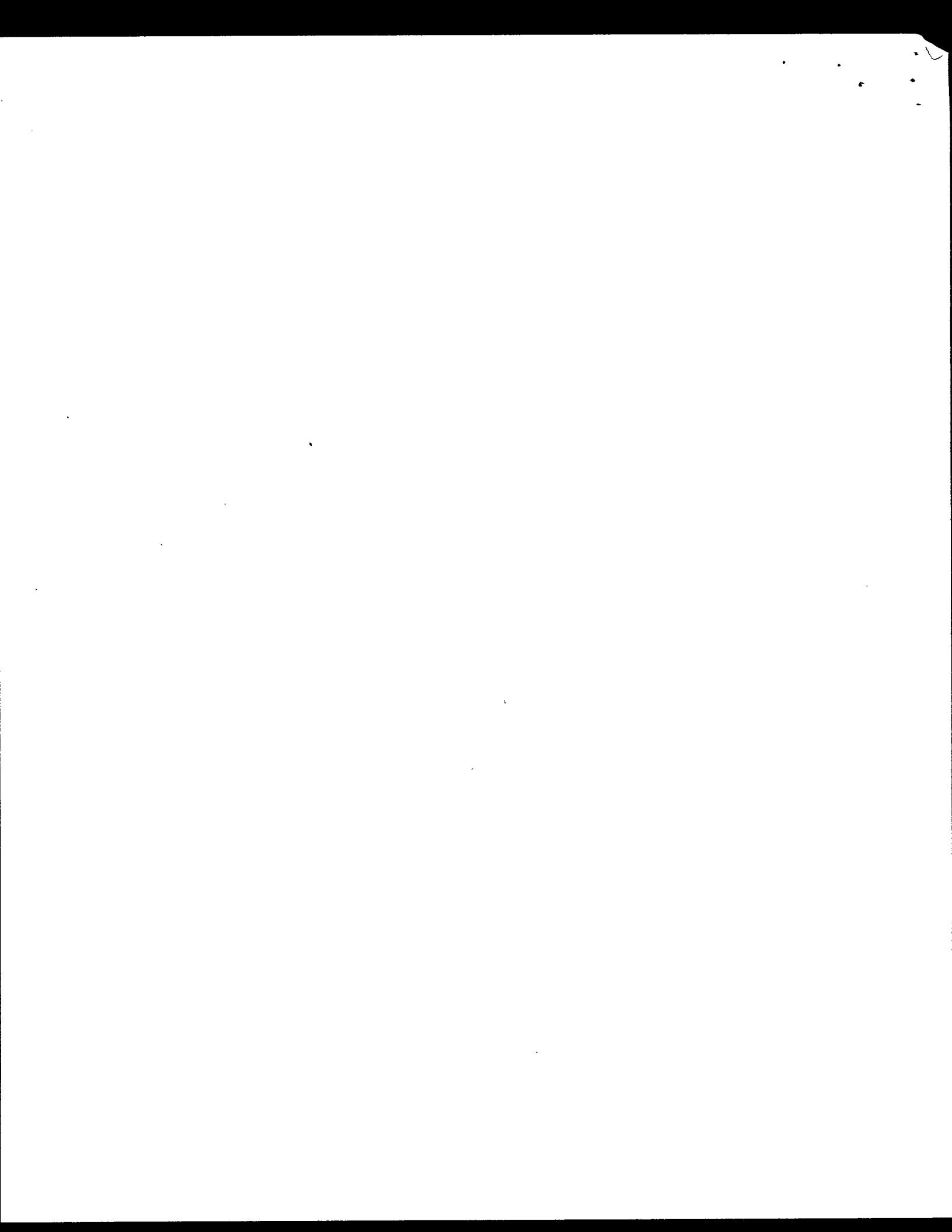


AvAlaska, Inc.
4340 Postmark Dr.
Document Number N57HY


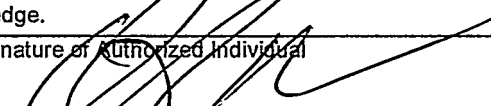
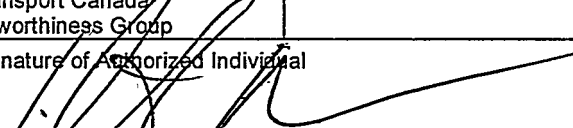
Airplane Flight Manual Supplement for
Sky International, Inc. (Aviat) Model Husky A-1B
N57HY; S/N 2057
With Oil cooler shutter Installed

IV. Performance Information: No Change

V. Loading Information: No Change



FAA

 <p>US Department of Transportation Federal Aviation Administration</p>		<p>MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)</p>		<p>Form Approved OMB No. 2120-0020</p>	
				<p>For FAA Use Only</p>	
				<p>Office Identification FAA-AAL-FSDO-03 <i>RSJK</i></p>	
<p>INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each violation (Section 901 of Federal Aviation Act of 1958).</p>					
1. Aircraft	Make	Aviat		Model A-1B	
	Serial No.	2065		Nationality and Registration Mark N67HY	
2. Owner	Name (As shown on registration certificate)			Address (As shown on registration certificate)	
	Buena Vista Aviation, Inc.			7911 Canal Street Anchorage, AK 99502	
3. For FAA Use Only					
4. Unit Identification					
				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	----- (As described in Item 1 above) -----				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				
6. Conformity Statement					
A. Agency Name and Address		B. Kind of Agency		C. Certificate No.	
Clifford D. Belleau 4340 POSTMARK DR. ANCHORAGE AK 99502		<input checked="" type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input type="checkbox"/> Certified Repair Station <input type="checkbox"/> Manufacturer		574-56-2293	
<p>D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.</p>					
Date		Signature of Authorized Individual			
6-27-00					
7. Approval for Return To Service					
<p>Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is <input checked="" type="checkbox"/> APPROVED <input type="checkbox"/> REJECTED</p>					
BY	FAA Flt. Standards Inspector	Manufacturer	<input checked="" type="checkbox"/>	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station		Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection		Certificate or Designation No.	Signature of Authorized Individual		
6-27-00		574-56-2293			

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.):

On N67HY:

1. Installed F. A. Dodge Rear Seat Heater kit part number 3175-1, kit is approved for installation on Piper PA-18 series aircraft under STC SA02094AK. Installation on this Aviat aircraft is IAW the multiple field approval of N57HY. See FAA form 337 dated June 26, 2000.
2. An Airplane Flight Manual Supplement, document number N57HY-2, dated June 26, 2000 has been installed in the aircraft Approved Flight Manual.
3. Aircraft weight and balance and equipment list has been revised to reflect installation.
4. **Additional instructions for continued airworthiness beyond 14 CFR 43:** (1) Inspect flapper in the control box for condition at annual inspection, and that flapper is making a seal in the closed position. (2) Insure adequate clearance between carburetor heat shroud and cowling after any removal and reinstallation of shroud.

-----end of entry-----

Additional Sheets are Attached

AvAlaska, Inc.
FAA Approved
Airplane Flight Manual Supplement
For
Aviat A-1B

Serial Number: ~~2057~~ 2065
N Number: ~~N57HY~~ N67HY

This supplement must be attached to the FAA Approved Airplane Flight Manual dated January 28, 1998, or later FAA approved revision, when F. A. Dodge Auxiliary Heater kit #3175 is installed in accordance with FAA Form 337 dated June 26, 2000. The information contained herein supplements or supersedes the information in the basic airplane flight manual only in those areas listed herein. For limitations, procedures and performance information not contained in this supplement, consult the basic airplane flight manual.

SECTION I. Operating Limitations:

G. Markings and Placards: The following information must be displayed in the form of a Placard(s) next to the oil cooler shutter control knob:

AUXILIARY HEAT -- PUSH TO CLOSE

SECTION III. Emergency Procedures:

D. FIRES

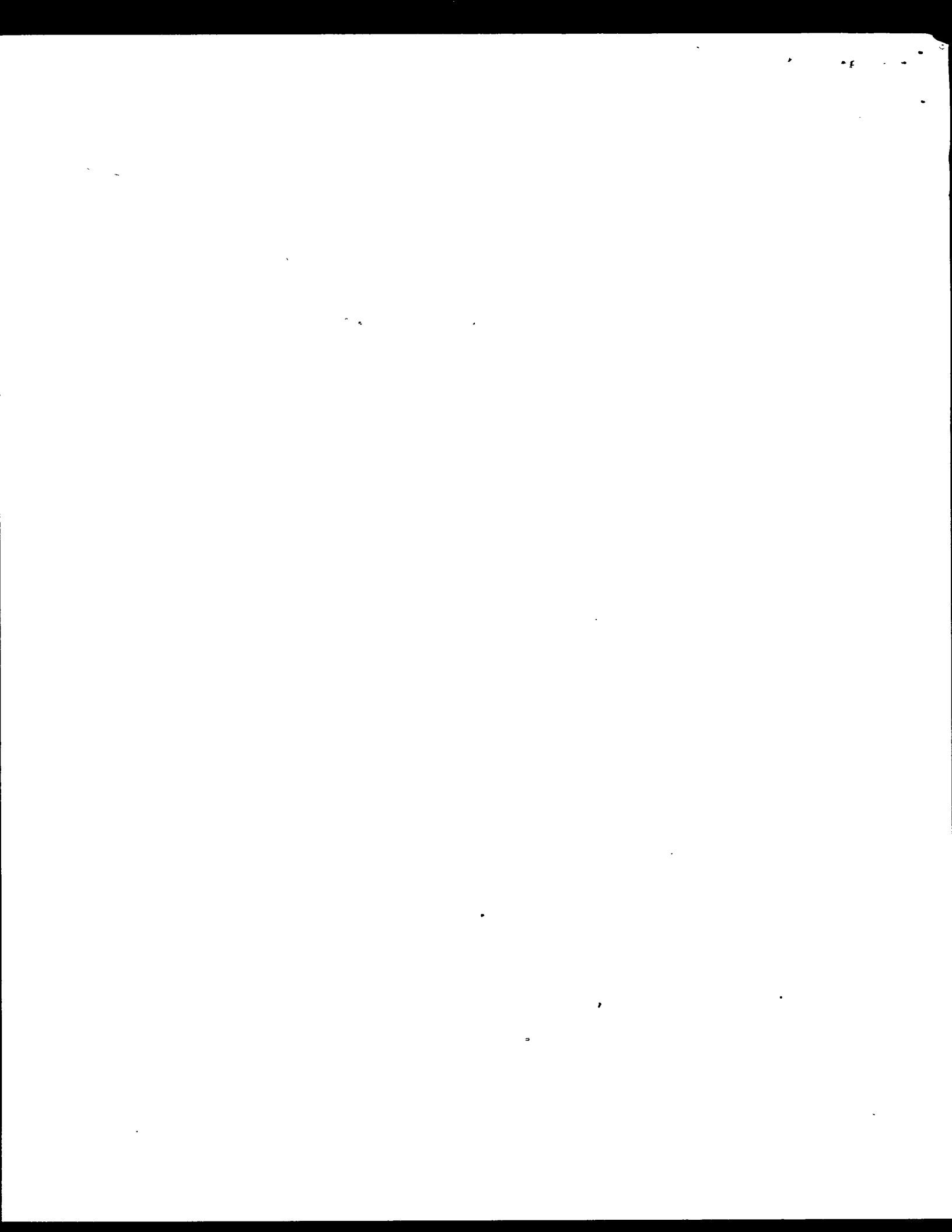
3. FIRE IN FLIGHT (ENGINE)

d(a). AUXILIARY HEATCLOSE

IV. Performance Information: No Change

V. Loading Information: No Change

FAA Approved: James P. Chudy
Manager, Aircraft Certification Office
Federal Aviation Administration
Anchorage, Alaska



Owner



US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only
Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each violation (Section 901 of Federal Aviation Act of 1958).

1. Aircraft	Make Aviat	Model A-1B
	Serial No. 2057	Nationality and Registration Mark N57HY
2. Owner	Name (As shown on registration certificate) Buena Vista Aviation, Inc.	Address (As shown on registration certificate) 7911 Canal Street Anchorage, AK 99502

3. For FAA Use Only

The alteration identified herein complies with the applicable requirements and is approved for duplication on identical aircraft Make, model and altered configuration when accomplished by the original modifier.

06/26/00 Date Signature FAA-AAL-FSDO-3

4. Unit Identification

5. Type

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency Name and Address Clifford D. Belleau 4340 POSTMARK DR. ANCHORAGE AK 99502	B. Kind of Agency	C. Certificate No. 574-56-2293
	<input checked="" type="checkbox"/> U.S. Certificated Mechanic	
	<input type="checkbox"/> Foreign Certificated Mechanic	
	<input type="checkbox"/> Certified Repair Station	
	<input type="checkbox"/> Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 6-26-00	Signature of Authorized Individual
-----------------	--

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

BY	FAA Fit. Standards Inspector	Manufacturer	<input checked="" type="checkbox"/>	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station		Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection 6-26-00		Certificate or Designation No. 574-56-2293	Signature of Authorized Individual 		

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished


(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.):

On N57HY:

1. Installed F. A. Dodge Rear Seat Heater kit part number 3175-1, kit is approved for installation on Piper PA-18 series aircraft under STC SA02094AK. Installation on this Aviat aircraft is as follows:
2. The heat valve is located on the lower firewall aft face approximately in the center and 18" above the floor. The heat valve is secured with four each MS 221044-N08 nuts, AN960-8L washers, and MS35206-246 machine screws. The heat valve is sealed upon installation with MIL5-38249 AMEND. 3 sealant.
3. The original factory carb heat duct was disconnected from the muffler shroud and a Cessna P/N 1750011-4 shroud was installed on the # 3 cylinder exhaust riser. Existing carburetor heat duct was connected to the Cessna shroud.
4. The Cessna shroud was modified by trimming the lower corner to allow the cowling to clear.
5. The heat control valve is located on the right side of the instrument panel adjacent to the standard cabin heat control, using the kit supplied mounting bracket and hardware. A Cessna P/N S1230-17 control actuates the valve. The control is placarded "Auxiliary Heat".
6. An additional 2" inlet was welded to the forward side of the standard muffler heat shroud. This inlet was connected to Aviat P/N 35624-011 screen, 35626-002 flange, and 35624-012 retainer mounted on the right hand side of lower cowling nose bowl with four each AN526C632 screws, MS21044N06 nuts, and AN960-6 washers. A 12" long section of 2" SCAT connects the shroud and inlet. The SCAT is secured to the nosebowl at the middle of the run by one each AN742D-34 clamp, AN525-832R8 screw, MS21044N08 nut to prevent fouling of the alternator.
7. An Airplane Flight Manual Supplement, document number N57HY-2, dated June 26, 2000 has been installed in the aircraft Approved Flight Manual.
8. Aircraft weight and balance and equipment list has been revised to reflect installation.
9. **Additional instructions for continued airworthiness beyond 14 CFR 43:** (1) Inspect flapper in the control box for condition at annual inspection, and that flapper is making a seal in the closed position. (2) Insure adequate clearance between carburetor heat shroud and cowling after any removal and reinstallation of shroud.

-----end of entry-----

FAA

 US Department of Transportation Federal Aviation Administration	MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)	Form Approved OMB No. 2120-0020
		For FAA Use Only
		Office Identification <i>FAA-AAL-FSDO-03-RBC</i>

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each violation (Section 901 of Federal Aviation Act of 1958).

1. Aircraft	Make Aviat	Model A-1B
	Serial No. 2065	Nationality and Registration Mark N67HY
2. Owner	Name (As shown on registration certificate) Buena Vista Aviation, Inc.	Address (As shown on registration certificate) 7911 Canal Street Anchorage, AK 99502

3. For FAA Use Only

The alteration identified herein complies with the applicable requirements and is approved for the above described aircraft, subject to conformity inspection by a person authorized in FAR Part 43.

06/27/00 Date *[Signature]* Signature FAA-AAL-FSDO-3

4. Unit Identification				5. Type	
Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	~~~~~ (As described in Item 1 above) ~~~~~				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency Name and Address AVALASKA, INC. 4340 POSTMARK DR. ANCHORAGE AK 99502	B. Kind of Agency <input type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input checked="" type="checkbox"/> Certified Repair Station <input type="checkbox"/> Manufacturer	C. Certificate No. V2AR414J
--	--	---------------------------------------

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date <i>6-29-00</i>	Signature of Authorized Individual <i>[Signature]</i>
------------------------	--

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

BY	FAA Flt. Standards Inspector		Manufacturer	Inspection Authorization	Other (Specify)
	FAA Designee	X	Repair Station	Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection <i>6-29-00</i>	Certificate or Designation No. V2AR414J	Signature of Authorized Individual <i>[Signature]</i>
---	--	--

NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished

(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.):

On N67HY:

1. ACK ELT E-01 antenna was bent aft approximately 4" above lower end. This allows the antenna to clear the aft baggage shelf.
2. This alteration is approved under STC SA02134AK.
3. No change to aircraft weight and balance or equipment list.
4. **No additional instructions for continued airworthiness beyond 14 CFR 43.**

-----end of entry-----

Additional Sheets are Attached



U.S. Department of Transportation
Federal Aviation Administration

APPLICATION FOR AIRWORTHINESS CERTIFICATE

INSTRUCTIONS - Print or type. Do not write in shaded areas; these are for FAA use only. Submit original only to an authorized FAA Representative. If additional space is required, use attachment. For special flight permits complete Sections II and VI as applicable.

I. AIRCRAFT DESCRIPTION	1. REGISTRATION MARK N67HY	2. AIRCRAFT BUILDER'S NAME (Make) Aviat Aircraft Inc.	3. AIRCRAFT MODEL DESIGNATION A-1B	4. YR. MFR. 2000	FAA CODING 2210406
	5. AIRCRAFT SERIAL NO. 2065	6. ENGINE BUILDER'S NAME (Make) Textron Lycoming	7. ENGINE MODEL DESIGNATION 0-360-A1P		41514
	8. NUMBER OF ENGINES 1	9. PROPELLER BUILDER'S NAME (Make) Hartzell	10. PROPELLER MODEL DESIGNATION HC-C2YK-1BF/F7666A		11. AIRCRAFT IS (Check if applicable) <input checked="" type="checkbox"/> IMPORT

II. CERTIFICATION REQUESTED	APPLICATION IF HEREBY MADE FOR: (Check applicable items)										
	<input checked="" type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	STANDARD AIRWORTHINESS CERTIFICATE (Indicate category)			<input checked="" type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>
	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	SPECIAL AIRWORTHINESS CERTIFICATE (Check appropriate items)			<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>
	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	2	LIMITED	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>
	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	5	PROVISIONAL (Indicate class)	1	CLASS I	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>
	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	3	RESTRICTED (Indicate operation(s) to be conducted)	2	CLASS II	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>
	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	4	EXPERIMENTAL (Indicate operation(s) to be conducted)	1	AGRICULTURE AND PEST CONTROL	2	AERIAL SURVEYING	3	AERIAL ADVERTISING
	<input type="checkbox"/>	<input type="checkbox"/>	4			FOREST (Wildlife conservation)	5	PATROLLING	6	WEATHER CONTROL	
	<input type="checkbox"/>	<input type="checkbox"/>	7			CARRIAGE OF CARGO	0	OTHER (Specify)			
	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	8	SPECIAL FLIGHT PERMIT (Indicate operation to be conducted, then complete Section VI or VII as applicable on reverse side)	1	RESEARCH AND DEVELOPMENT	2	AMATEUR BUILT	3	EXHIBITION
<input type="checkbox"/>	<input type="checkbox"/>	4	RACING			5	CREW TRAINING	<input type="checkbox"/>	MKT SURVEY		
<input type="checkbox"/>	<input type="checkbox"/>	0	TO SHOW COMPLIANCE WITH FAR								
<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	1	FERRY FLIGHT FOR REPAIRS, ALTERATIONS, MAINTENANCE OR STORAGE	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	
<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	2	EVACUATE FROM AREA OF IMPENDING DANGER	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	
<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	3	OPERATION IN EXCESS OF MAXIMUM CERTIFICATED TAKE-OFF WEIGHT	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	
<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	4	DELIVERING OR EXPORT	5	PRODUCTION FLIGHT TESTING	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	
<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	6	CUSTOMER DEMONSTRATION FLIGHTS	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	
<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	6	MULTIPLE AIRWORTHINESS CERTIFICATE (Check ABOVE "Restricted Operation" and "Standard" or "Limited" as applicable.)	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>	

III. OWNER'S CERTIFICATION	A. REGISTERED OWNER (As shown on certificate of aircraft registration)									
	NAME Aviat Aircraft Inc.					ADDRESS Box 1240 Afton, WY 83110				
	IF DEALER, CHECK HERE <input type="checkbox"/> X									
	B. AIRCRAFT CERTIFICATION BASIS (Check applicable blocks and complete items as indicated)									
<input checked="" type="checkbox"/>	AIRCRAFT SPECIFICATION OR TYPE CERTIFICATE DATA SHEET (Give No. and Revision No.) A22NM Rev 11				<input checked="" type="checkbox"/>	AIRWORTHINESS DIRECTIVES (Check if all applicable AD's complied with and give latest AD no.) 2000-04				
<input type="checkbox"/>	AIRCRAFT LISTING (Give page number(s)) N/A				<input type="checkbox"/>	SUPPLEMENTAL TYPE CERTIFICATE (List number of each STC incorporated) N/A				
C. AIRCRAFT OPERATION AND MAINTENANCE RECORDS										
<input checked="" type="checkbox"/>	CHECK IF RECORDS IN COMPLIANCE WITH FAR 91.173. 91.417		TOTAL AIRFRAME HOURS 1.9			EXPERIMENTAL ONLY (Enter hours flown since last certificate issued or renewed) 0.0				
D. CERTIFICATION - I hereby certify that I am the registered owner (or his agent) of the aircraft described above, that the aircraft is registered with the Federal Aviation Administration in accordance with Section 501 of the Federal Aviation Act of 1958, and applicable Federal Aviation Regulations, and that the aircraft has been inspected and is airworthy and eligible for the airworthiness certificate requested.										
DATE OF APPLICATION 3/7/2000			NAME AND TITLE (Print or type) Danny Hiner Manager				SIGNATURE <i>Danny Hiner</i>			

IV. INSPECTION AGENCY VERIFICATION	A. THE AIRCRAFT DESCRIBED ABOVE HAS BEEN INSPECTED AND FOUND AIRWORTHY BY (Complete this section only if FAR 21.183(d) applies).									
	2	FAR PART 121 or 127 CERTIFICATE HOLDER (Give Certificate No.)	3	CERTIFICATED MECHANIC (Give Certificate No.)	6	CERTIFICATED REPAIR STATION (Give Certificate No.)				
	2	AIRCRAFT MANUFACTURER (Give name or firm)								
DATE			TITLE				SIGNATURE			

V. FAA REPRESENTATIVE CERTIFICATION	(Check ALL applicable blocks in items A and B)											
	A. I find that the aircraft described in Section I or VII meets requirements for				<input checked="" type="checkbox"/>	THE CERTIFICATE REQUESTED						
	B. Inspection for a special flight permit under Section VII was conducted by				<input type="checkbox"/>	AMENDMENT OR MODIFICATION OF CURRENT AIRWORTHINESS CERTIFICATE						
					<input type="checkbox"/>	FAA INSPECTOR		FAR DESIGNEE				
				<input type="checkbox"/>	CERTIFICATE HOLDER UNDER		FAR 65		FAR 121, 127 OR 135		FAR 145	
DATE 3-7-2000		DISTRICT OFFICE SEA-MIDO		DESIGNEE'S SIGNATURE AND NO. <i>Albert G Humbert</i> 4				FAA INSPECTOR'S SIGNATURE 1				
										DMIR602538		

VI. PRODUCTION FLIGHT TESTING

A. MANUFACTURER		
NAME	ADDRESS	
B. PRODUCTION BASIS (Check applicable item)		
<input type="checkbox"/>	PRODUCTION CERTIFICATE (Give production certificate number)	
<input type="checkbox"/>	TYPE CERTIFICATE ONLY	
<input type="checkbox"/>	APPROVED PRODUCTION INSPECTION SYSTEM	
C. GIVE QUANTITY OF CERTIFICATES REQUIRED FOR OPERATING NEEDS —————→		
DATE OF APPLICATION	NAME AND TITLE (Print or type)	SIGNATURE

VII. SPECIAL FLIGHT PERMIT PURPOSES OTHER THAN PRODUCTION FLIGHT TEST

A. DESCRIPTION OF AIRCRAFT		
REGISTERED OWNER	ADDRESS	
BUILDER (Make)	MODEL	
SERIAL NUMBER	REGISTRATION MARK	
B. DESCRIPTION OF FLIGHT CUSTOMER DEMONSTRATION FLIGHTS <input type="checkbox"/> (Check if applicable)		
FROM	TO	
VIA	DEPARTURE DATE	DURATION
C. CREW REQUIRED TO OPERATE THE AIRCRAFT AND ITS EQUIPMENT		
<input type="checkbox"/>	<input type="checkbox"/>	<input type="checkbox"/>
PILOT	CO-PILOT	NAVIGATOR
OTHER (Specify)		
D. THE AIRCRAFT DOES NOT MEET THE APPLICABLE AIRWORTHINESS REQUIREMENTS AS FOLLOWS		
E. THE FOLLOWING RESTRICTIONS ARE CONSIDERED NECESSARY FOR SAFE OPERATION (Use attachment if necessary)		

F. CERTIFICATION - I hereby certify that I am the registered owner (or his agent) of the aircraft described above, that the aircraft is registered with the Federal Aviation Administration in accordance with Section 501 of the Federal Aviation Act of 1958, and applicable Federal Aviation Regulations; and that the aircraft has been inspected and is airworthy for the flight described.

DATE	NAME AND TITLE (Print or type)	SIGNATURE
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VIII. AIRWORTHINESS DOCUMENTATION (FAA use only)

<input checked="" type="checkbox"/>	A. Operating Limitations and Markings in Compliance with FAR 91.31 as Applicable 91.9	G. Statement of Conformity, FAA Form 8130-9 (Attach when required)
	B. Current Operating Limitations Attached	H. Foreign Airworthiness Certification for Import Aircraft (Attach when required)
	C. Data, Drawings, Photographs, etc. (Attach when required)	I. Previous Airworthiness Certificate Issued in Accordance with
<input checked="" type="checkbox"/>	D. Current Weight and Balance Information Available in Aircraft	FAR _____ CAR _____ (Original Attached)
	E. Major Repair and Alteration, FAA Form 337 (Attach when required)	J. Current Airworthiness Certificate Issued in Accordance with
<input checked="" type="checkbox"/>	F. This inspection Recorded in Aircraft Records	FAA Form 21.183 (a) (Copy attached)

UNITED STATES OF AMERICA
DEPARTMENT OF TRANSPORTATION - FEDERAL AVIATION ADMINISTRATION

COPY

STANDARD AIRWORTHINESS CERTIFICATE

1 NATIONALITY AND REGISTRATION MARKS N67HY	2. MANUFACTURER AND MODEL Aviat Aircraft Inc. A-1B	3. AIRCRAFT SERIAL NUMBER 2065	4 CATEGORY Normal
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5 AUTHORITY AND BASIS FOR ISSUANCE

This airworthiness certificate is issued pursuant to the Federal Aviation Act of 1958 and certifies that, as of the date of issuance, the aircraft to which issued has been inspected and found to conform to the type certificate therefor, to be in condition for safe operation, and has been shown to meet the requirements of the applicable comprehensive and detailed airworthiness code as provided by Annex 8 to the Convention on International Civil Aviation, except as noted herein.

Exceptions:

None

6 TERMS AND CONDITIONS

Unless sooner surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator, this airworthiness certificate is effective as long as the maintenance, preventative maintenance, and alterations are performed in accordance with Parts 21, 43, and 91 of the Federal Aviation Regulations, as appropriate, and the aircraft is registered in the United States.

DATE OF ISSUANCE 3-7-2000	FAA REPRESENTATIVE <i>Albert G Humbert</i> Albert G Humbert	DESIGNATION NUMBER DMIR602538NM
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Any alteration, reproduction, or misuse of this certificate may be punishable by a fine not exceeding \$1,000, or imprisonment not exceeding 3 years, or both. THIS CERTIFICATE MUST BE DISPLAYED IN THE AIRCRAFT IN ACCORDANCE WITH APPLICABLE FEDERAL AVIATION REGULATIONS.





US Department of Transportation
Federal Aviation Administration

MAJOR REPAIR AND ALTERATION
(Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020

For FAA Use Only

Office Identification

ANM FSD-04 JAC

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make	HUSKY	Model	A1-B
	Serial No.	2065	Nationality and Registration Mark	N67HY
2. Owner	Name (As shown on registration certificate)		Address (As shown on registration certificate)	
	AVIAT ACFT. INC.		672 SO. WASHINGTON AFTON WY. 83110	

3. For FAA Use Only

The data identification on this form was with the applicable airworthiness requirements and is approved for the above described aircraft, subject to conformity inspection by a person authorized in FAR Section 43.7*

Jan A. Clark
FAA Insp. ANM 04 Date 03-02-00

4. Unit Identification

Unit	Make	Model	Serial No.	5. Type	
				Repair	Alteration
AIRFRAME	(As described in Item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

6. Conformity Statement

A. Agency's Name and Address AVIAT ACFT. INC. 672 SO. WASHINGTON AFTON WY. 83110	B. Kind of Agency	C. Certificate No. 704NM
	<input type="checkbox"/> U.S. Certificated Mechanic	
	<input type="checkbox"/> Foreign Certificated Mechanic	
	<input checked="" type="checkbox"/> Manufacturer	

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date	2-28-00	Signature of Authorized Individual	STEVE SABAT <i>Steve Sabat</i>
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7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED

BY	FAA Fit Standards Inspector	Manufacturer	X	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station		Person Approved by Transport Canada Airworthiness Group	

Date of Approval or Rejection	3-17-00	Certificate or Designation No.	520906239	Signature of Authorized Individual	<i>Robert Mortenson</i>
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1-1-1
1-1-1

(UNIT)	(TSO)	(MANUAL)	(CIRC. PROT.)
GNC 250XL	C37d CLASS 4 C38d CLASS C	GARMIN PN-190-00076-62 REV. B	10A
KMA 24H	C50c CLASS 1	KING PN-006-00586-0002 REV. 2	4A
AK-550-6	C71	AMERIKING DOCUMENT # IM-550	15A
PULSELITE 1200 SERIES	NA	MANUF. STC.SA4005NM DRAWING # 220P0001	3A
AVIATION ELEC.INC. TIB 2000	NA	NA	1A
ACK E-01 ELT	C91a	E-01 ELT INST. AND OPERATION MANUAL, REV.8-12-98	NA
GA 56 GPS ANT.	C129a CLASS A1	GARMIN PN-190-00094-00, REV,B	NA
TFM 138B	NA	TIL DOCUMENT #97RE221, REV.A	3A
COMANT CI-177 FM ANTENNA	C37b, C38b, DO138	COMANT PN-D12101	NA

WEIGHT AND BALANCE. AND EQUIPT. LIST HAS BEEN AMENDED.

-----END-----

INSTRUCTIONS FOR CONTINUED AIRWORTHINESS:
REFER TO COMPONENT MANUFACTURERS MANUALS LISTED ABOVE.

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